

Configuration Flatness of Lagrangian Systems Underactuated by One Control

Muruhan Rathinam* and Richard M. Murray

Division of Engineering and Applied Science

California Institute of Technology

Pasadena, CA 91125

muruhan@ama.caltech.edu, murray@indra.caltech.edu

5-Aug-96

Abstract

Lagrangian control systems that are differentially flat with flat outputs that only depend on configuration variables are said to be configuration flat. We provide a complete characterisation of configuration flatness for systems with n degrees of freedom and $n - 1$ controls whose range of control forces only depends on configuration and whose Lagrangian has the form of kinetic energy minus potential. The method presented allows us to determine if such a system is configuration flat and, if so provides a constructive method for finding all possible configuration flat outputs. Our characterisation relates configuration flatness to Riemannian geometry.

1. Introduction

Roughly speaking an underdetermined system of ordinary differential equations

$$F^k(t, x^1, \dots, x^N, \dot{x}^1, \dots, \dot{x}^N) = 0, \quad k = 1, \dots, n < N$$

is differentially flat if there is a smooth locally 1-1 correspondence between solutions $x(t)$ of the system and arbitrary functions $y(t)$, of the form

$$\begin{aligned} x(t) &= g(t, y(t), \dots, y^{(l)}(t)), \\ y(t) &= h(t, x(t), \dots, x^{(q)}(t)), \end{aligned}$$

where $(y^1, \dots, y^p) \in \mathbb{R}^p$ ($p = N - n$). Here g, h are smooth maps, $y^{(k)}$ is the k^{th} derivative of y and l, q are integers. The variables y^j are referred to as flat outputs. The special class of systems given by

$$x^i = f^i(t, x^1, \dots, x^n, u^1, \dots, u^p), \quad i = 1, \dots, n$$

are more familiar to control theorists and the flat outputs depend on states, inputs, and derivatives of inputs

$$y^j = h^j(t, x, u, u^{(1)}, \dots, u^{(q)}), \quad j = 1, \dots, p.$$

For a detailed discussion of differential flatness see Fliess et al. [1, 2], Martin [3], Pomet [4], van Nieuwstadt et al. [5] and Rathinam and Sluis [6].

*Research supported in part by NSF Grant CMS-9502224.

The importance of flatness to control applications lies in the fact that it provides a systematic and relatively simple way to generate solution trajectories between two given states. One uses the maps g and h to transform between original system space (states as well as inputs) and the smaller dimensional flat output space. See van Nieuwstadt and Murray [7] and Murray et al. [8] for more details.

In the case of single input systems a complete characterization of differential flatness is available, see e.g. Shadwick [9]. In that case, flatness is the same as feedback linearizability. In the framework of exterior differential systems, checking for flatness reduces to calculating “derived systems” and checking certain rank and integrability conditions. See [5, 10]. For multi-input systems no complete theory exists.

Many interesting examples of mechanical systems are differentially flat and in most known examples flat outputs have been found that depend only on the configuration variables but not on their derivatives. We refer to such flat outputs as “configuration flat outputs” and systems possessing such outputs as “configuration flat”. All Lagrangian systems that are fully actuated (number of controls equals number of degrees of freedom) are configuration flat with all the configuration variables as flat outputs. See [8] for a catalogue of other examples. In this paper we completely characterise configuration flatness for a special class of mechanical systems. The class under consideration involves systems whose dynamics is described by Lagrangian mechanics with a Lagrangian function of the form “kinetic energy minus potential”. Also the number of independent controls is assumed to be one less than the number of degrees of freedom (the simplest case next to fully actuated systems) and the possible range of control forces only depends on the configuration and not on the velocity. We describe an algorithm for deciding if such a system is configuration flat and if it is so, we describe a procedure for finding all possible configuration flat outputs.

The paper is organized as follows. Section 2 introduces some concepts from Lagrangian control systems theory and also provides a definition of configuration flatness. Section 3 introduces some concepts from Rie-

mannian geometry that are necessary for our theory and also states and proves the main proposition and also outlines an algorithm for coordinate calculations to check configuration flatness. Finally Section 4 gives two examples to illustrate the methodology.

2. Lagrangian control systems and configuration flatness

Consider a Lagrangian system with configuration manifold Q of dimension n and a Lagrangian $L : TQ \rightarrow \mathbb{R}$. When no external (generalised) forces are applied, the motion of this system satisfies the Euler-Lagrange equations, written in coordinates (q^1, \dots, q^n) as

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}^i} \right) - \frac{\partial L}{\partial q^i} = 0, \quad i = 1, \dots, n. \quad (1)$$

In a control situation external control forces are applied and it is natural to think of forces as covectors on the manifold Q . In other words, for a configuration $q \in Q$ the total external force acting on the system can be represented by an element of T_q^*Q . This is because forces naturally pair with velocities, which can be thought of as elements of T_qQ , to give instantaneous power. The possible range of control forces lies in a subspace of T_q^*Q which may depend on position q as well as velocity v_q . In other words the control forces can be described by a horizontal valued codistribution $\bar{P} \subset T^*(TQ)$, and $p = \dim \bar{P}$ is the number of independent controls. For an interesting and wide class of systems this subspace only depends on configuration q and hence can be described by a codistribution $P \subset T^*Q$ of dimension p . For the rest of the discussion we shall only consider this case. All feasible paths (solutions) of such a system are characterised by the following underdetermined system of ODEs in coordinates (q^1, \dots, q^n) :

$$a_k^i \left(\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}^i} \right) - \frac{\partial L}{\partial q^i} \right) = 0, \quad k = 1, \dots, n - p \quad (2)$$

where $a_k^i \frac{\partial}{\partial q^i}$ for $k = 1, \dots, n - p$ span the annihilator of P , denoted $\text{ann } P$.

It is useful to think in terms of the associated submanifold $\mathcal{E} \subset J^2(\mathbb{R}, Q)$ of the second order jet space (see [11]), which geometrically describes such a second order system of equations. \mathcal{E} has codimension $n - p$ and in local coordinates $(t, q, \dot{q}, \ddot{q})$ is cut out by the common zeroes of the functions

$$a_k^i \left(\frac{\partial^2 L}{\partial \dot{q}^i \partial \dot{q}^j} \ddot{q}^j + \frac{\partial^2 L}{\partial \dot{q}^i \partial q^j} \dot{q}^j - \frac{\partial L}{\partial q^i} \right), \quad k = 1, \dots, n - p.$$

Let $q \in Q$ be a point and let $y : U \subset Q \rightarrow \mathbb{R}^p$ be a submersion locally defined around q . Let $y = (y^1, \dots, y^p)$. We say y^1, \dots, y^p are *differentially independent* around q if y^1, \dots, y^p do not have to satisfy an ODE along solutions local to q . More precisely, when restricted to \mathcal{E} , $dy^1, \dots, dy^p, d\dot{y}^1, \dots, d\dot{y}^p, d\ddot{y}^1, \dots, d\ddot{y}^p$ are linearly independent for generic points on $\pi_2^{-1}(V) \cap \mathcal{E}$ where $V \subset U$ is an open neighbourhood of q and

$\pi_2 : J^2(\mathbb{R}, Q) \rightarrow Q$ is the standard projection. If $dy^1, \dots, dy^p, d\dot{y}^1, \dots, d\dot{y}^p, d\ddot{y}^1, \dots, d\ddot{y}^p$ are linearly dependent when restricted to \mathcal{E} , for points on $\pi_2^{-1}(V) \cap \mathcal{E}$ where $V \subset U$ is an open neighbourhood of q then y^1, \dots, y^p are *differentially dependent* around q .

Suppose y^1, \dots, y^p are differentially independent around q . If there are functions f^i and a neighbourhood W of q such that along a generic solution $c : \mathbb{R} \rightarrow W \subset Q$,

$$(z^i \circ c)(t) = f^i((y \circ c)(t), \dots, \frac{d^r}{dt^r}(y \circ c)(t)), \quad (3)$$

$$i = 1, \dots, n - p$$

where z^1, \dots, z^{n-p} are any complementary coordinates to y^1, \dots, y^p , then y^1, \dots, y^p are said to be *configuration flat outputs* around q and the system is *configuration flat* around q . In other words, given $y^1(t), \dots, y^p(t)$ we can determine a (locally) unique trajectory for the Lagrangian system (2).

We present the following lemma which will be of use later.

Lemma 1 Let $q \in Q$, U an open neighbourhood of q , and $y : U \rightarrow \mathbb{R}^p$ be a configuration flat output. Then generically the set of solutions $c : \mathbb{R} \rightarrow U$ that project down to the same curve $y \circ c$ are all isolated.

Proof By definition of flatness along generic solutions, given $y(t)$ the complementary coordinates $z(t)$ are locally uniquely determined by equations (3). ■

3. Mechanical systems with n degrees of freedom and $n - 1$ controls

Consider the mechanical system whose Lagrangian is given by

$$L(v) = \frac{1}{2}g(v, v) - V \circ \tau_Q(v), \quad (4)$$

where g is the Riemannian metric (assumed to be non-degenerate) corresponding to kinetic energy and V is the potential energy function on Q and $\tau_Q : TQ \rightarrow Q$ is the tangent bundle projection. Suppose the number of controls $p = n - 1$, in other words $\dim P = n - 1$. In this section we shall present a method for determining if this system is configuration flat. If the system is configuration flat our approach provides us with a constructive method for finding all possible (configuration) flat outputs.

Before proceeding further we present some concepts from Riemannian geometry. Given a metric g we have a notion of differentiation of objects on the manifold such as functions, vector fields, differential forms and tensors along a given vector field Z . This is the covariant derivative ∇ given by the Levi-Civita connection (see [12]). ∇_Z denotes covariant derivative along a vectorfield Z and is related to parallel (with respect to metric) transport of objects along the integral curves of Z . The covariant derivative of a function f along Z

denoted $\nabla_Z f$ is just the familiar directional derivative $Z(f)$ or the Lie derivative. But the covariant derivative of a vectorfield X along Z denoted $\nabla_Z X$ is not the same as the Lie derivative $[Z, X]$. Some properties of ∇ are

$$\nabla_Z(X_1 + X_2) = \nabla_Z X_1 + \nabla_Z X_2 \quad (5)$$

$$\nabla_Z(fX) = \nabla_Z X + Z(f)X \quad (6)$$

$$\nabla_{fZ} X = f\nabla_Z X \quad (7)$$

$$\nabla_Z X - \nabla_X Z = [Z, X] \quad (8)$$

where X, X_1, X_2, Z are arbitrary vector fields and f is an arbitrary function on the manifold.

In a coordinate system (q^1, \dots, q^n) on manifold Q the covariant derivatives are calculated with the aid of Christoffel symbols Γ_{jk}^i , where $i, j, k = 1, \dots, n$ and Christoffel symbols are defined by

$$\nabla_{\frac{\partial}{\partial q^j}} \frac{\partial}{\partial q^k} = \Gamma_{jk}^i \frac{\partial}{\partial q^i}. \quad (9)$$

From the properties (8) of ∇ it follows that $\Gamma_{jk}^i = \Gamma_{kj}^i$. Γ_{jk}^i can be computed from metric g by the formula

$$\Gamma_{jk}^m = \frac{1}{2} \left(\frac{\partial g_{ik}}{\partial q^j} + \frac{\partial g_{ij}}{\partial q^k} - \frac{\partial g_{jk}}{\partial q^i} \right) g^{im}, \quad j, m = 1, \dots, n, \quad (10)$$

where $g^{ik} g_{kj} = \delta_j^i$ (g^{ik} are components of the inverse of matrix g_{ik}). Then the covariant derivative of vectorfield $X = X^k \frac{\partial}{\partial q^k}$ along $Z = Z^j \frac{\partial}{\partial q^j}$ is given by

$$\nabla_Z X = Z^j X^k \Gamma_{jk}^i \frac{\partial}{\partial q^i} + Z^j \frac{\partial X^k}{\partial q^j} \frac{\partial}{\partial q^k}. \quad (11)$$

For the mechanical system under consideration let us define an associated distribution D by

$$D = \text{span}\{\xi, \nabla_Z \xi : Z \in \mathfrak{X}(Q)\}, \quad (12)$$

where ξ is any vector field such that $\text{ann } P = \text{span}\{\xi\}$ and $\mathfrak{X}(Q)$ is the set of all smooth vector fields on Q .

It is easy to check that D doesn't depend on the choice of $\xi \in \text{ann } P$. By the linearity of covariant derivative it follows that

$$D = \text{span}\{\xi, \nabla_{\frac{\partial}{\partial q^i}} \xi : i = 1, \dots, n\} \quad (13)$$

where (q^1, \dots, q^n) are any set of coordinates. Hence D is easily calculated using equations (10), (11) and (13). The following proposition characterises configuration flat outputs y^1, \dots, y^p by conditions on $\ker Ty$, which in coordinates is the null space of the Jacobian of the map y .

Proposition 2 Let q be a point on Q , U an open neighbourhood of q and suppose $y : U \subset Q \rightarrow \mathbb{R}^p$ is a submersion. If y^1, \dots, y^p are configuration flat outputs, then

$$g(\ker Ty, D) = 0. \quad (14)$$

Conversely if $g(\ker Ty, D) = 0$ and if certain regularity condition holds at q then y^1, \dots, y^p are configuration flat outputs around q .

The regularity condition is that the ratios of functions in the following set should not all be the same at q :

$$\{\nabla_\eta(g(\xi, Z)) : g(\xi, Z), \nabla_\eta(g(\nabla_{Z_1} Z_2, \xi)) : g(\nabla_{Z_1} Z_2, \xi), \quad (15)$$

$$\nabla_\eta(\xi(V)) : \xi(V)\},$$

where Z, Z_1, Z_2 are arbitrary vector fields around q that are y -related to some vectorfield on \mathbb{R}^p and ξ, η are fixed nonvanishing vector fields such that $\text{ann } P = \text{span}\{\xi\}$ and $\ker Ty = \text{span}\{\eta\}$.

Remark 3 Proposition 2 states the conditions for configuration flatness in intrinsic geometric terms. In coordinates the algorithm for deciding if the system is configuration flat is as follows. Calculate D using equation (13). If $D = TQ$ then system is not configuration flat, since for any y , one can find some vector field $Z \in D = TQ$, such that $g(\ker Ty, Z) \neq 0$. Suppose $\dim D \leq n - 1$. Then choose a one dimensional distribution, say spanned by a vectorfield η , that is orthogonal to D . Since a one dimensional distribution is integrable locally, one can find independent functions y^1, \dots, y^p ($p = n - 1$) around q that "cut out" the leaves of the corresponding foliation. These will be flat outputs provided the regularity conditions are met.

The regularity conditions can be checked in coordinates as follows. Choose a function z that completes y^1, \dots, y^p to a coordinate system. Then y^1, \dots, y^p will be flat outputs if the following ratios of functions are not all identically equal in a local neighbourhood:

$$\begin{aligned} \frac{\partial}{\partial z} \left(g\left(\xi, \frac{\partial}{\partial y^j}\right) \right) : g\left(\xi, \frac{\partial}{\partial y^j}\right), \quad j = 1, \dots, p \\ \frac{\partial}{\partial z} \left(g\left(\nabla_{\frac{\partial}{\partial y^k}} \frac{\partial}{\partial y^j}, \xi\right) \right) : g\left(\nabla_{\frac{\partial}{\partial y^k}} \frac{\partial}{\partial y^j}, \xi\right), \quad j, k = 1, \dots, p \end{aligned} \quad (16)$$

$$\frac{\partial}{\partial z} (\xi(V)) : \xi(V).$$

If these are all identically equal that means y^1, \dots, y^p are differentially dependent and another one dimensional distribution must be tried.

Proof (of Proposition 2) : Given a submersion $y : Q \rightarrow \mathbb{R}^p$, one can choose a local coordinate chart on Q such that y is the canonical submersion of \mathbb{R}^n onto \mathbb{R}^p . Let the corresponding coordinates on Q be (q^1, \dots, q^n) . Then, $y^j(q) = q^j$ for $j = 1, \dots, p = n - 1$. Let $\xi = a^i \frac{\partial}{\partial q^i}$ span $\text{ann } P$. Then all solutions of the system satisfy the single ODE

$$a^i \left(\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}^i} \right) - \frac{\partial L}{\partial q^i} \right) = 0. \quad (17)$$

Suppose in these coordinates g is given by g_{ij} . Then we can rewrite equation (17) as

$$a^i \left(g_{ij} \ddot{q}^j + \frac{\partial g_{ik}}{\partial q^j} \dot{q}^j \dot{q}^k - \frac{1}{2} \frac{\partial g_{jk}}{\partial q^i} \dot{q}^j \dot{q}^k + \frac{\partial V}{\partial q^i} \right) = 0. \quad (18)$$

Using the formula (10) for the Christoffel symbols and using $q^j = y^j$ for $j = 1, \dots, p$ to separate the terms involving \dot{q}^n and \ddot{q}^n , we rewrite equation (18) as,

$$\begin{aligned} a^i (g_{ij} \ddot{y}^j + \Gamma_{jk}^m g_{mi} \dot{y}^j \dot{y}^k + \frac{\partial V}{\partial q^i} + g_{in} \ddot{q}^n + \Gamma_{nn}^m g_{mi} (\dot{q}^n)^2 \\ + \Gamma_{jn}^m g_{mi} \dot{y}^j \dot{q}^n) = 0 \end{aligned} \quad (19)$$

where range of summation of various indices is clear.

Necessity: Suppose y are flat outputs. Then it follows that the coefficient of \ddot{q}^n in the above ODE must be zero. Otherwise we can rewrite the equation as

$$\frac{d\dot{q}^n}{dt} = f(y, \dot{y}, \ddot{y}, q^n, \dot{q}^n)$$

for some smooth function f , and by existence theorem of solutions to ODEs, given any curve $y(t)$ we get a 2-parameter family of solutions $q(t)$ (parametrised by initial conditions $q^n(t_0), \dot{q}^n(t_0)$) that project to $y(t)$ and they are not isolated from each other and hence by Lemma 1 y cannot be flat, contradicting our assumption. So $a^i g_{in} = 0$ and this leaves us with an ODE of the form

$$A(y)(\dot{q}^n)^2 + B(y, \dot{y})\dot{q}^n + C(y, \dot{y}, \ddot{y}, q^n) = 0.$$

A similar reasoning tells us that the term \dot{q}^n should be absent, in other words $A(y) = 0$ and $B(y, \dot{y}) = 0$. Here A and B are given by,

$$A = a^i \Gamma_{nn}^m g_{mi} \quad B = a^i \Gamma_{jn}^m g_{mi} \dot{y}^j.$$

Observe that B is linear in terms \dot{y} with coefficients that are functions only of (y, q^n) . Hence the condition $B = 0$ can be written as $n - 1$ equations that set the coefficients of \dot{y}^j to be zero. The equation $A = 0$ has the same form as these, and we get the following n equations:

$$a^i \Gamma_{jn}^m g_{im} = 0, \quad j = 1, \dots, n.$$

So all together flatness of y implies the following equations,

$$\begin{aligned} a^i g_{in} &= 0 \\ a^i \Gamma_{jn}^m g_{im} &= 0, \quad j = 1, \dots, n. \end{aligned} \quad (20)$$

If $\ker Ty = \text{span}\{\eta\}$, then in our choice of coordinates $\eta = \lambda \frac{\partial}{\partial q^n}$ where λ is some nonvanishing function on Q . Hence, $g(\xi, \eta) = a^i g_{in} = 0$ by the first condition, where $\xi = a^i \frac{\partial}{\partial q^i}$ spans $\text{ann } P$. Also since

$$\nabla_{\frac{\partial}{\partial q^j}} \eta = \lambda \Gamma_{jn}^m \frac{\partial}{\partial q^m} + \frac{\partial \lambda}{\partial q^j} \frac{\partial}{\partial q^n},$$

it follows that

$$g(\nabla_{\frac{\partial}{\partial q^j}} \eta, \xi) = \lambda a^i \Gamma_{jn}^m g_{im} + \frac{\partial \lambda}{\partial q^j} a^i g_{in} = 0.$$

But, by derivation property,

$$\nabla_Z (g(\xi, \eta)) = (\nabla_Z g)(\xi, \eta) + g(\nabla_Z \xi, \eta) + g(\xi, \nabla_Z \eta)$$

and since $\nabla_Z g = 0$ for any $Z \in \mathfrak{X}(Q)$ (by the property of Levi-Civita connection) and since $g(\eta, \xi) = 0$ it follows that

$$g(\nabla_{\frac{\partial}{\partial q^j}} \xi, \eta) = 0, \quad j = 1, \dots, n.$$

By linearity of ∇ it follows that

$$g(\nabla_Z \xi, \eta) = 0, \quad \forall Z \in \mathfrak{X}(Q).$$

Hence, $\ker Ty$ is orthogonal to D .

Sufficiency: Conversely, if $\ker Ty$ is orthogonal to D , previous reasoning shows that, in the same coordinate system the equations (20) hold. As seen before these imply that the solution curves of the system are given by the ODE

$$E(q^n, y, \dot{y}, \ddot{y}) = 0,$$

where

$$E = a^i g_{ij} \ddot{y}^j + a^i g_{im} \Gamma_{jk}^m \dot{y}^j \dot{y}^k + a^i \frac{\partial V}{\partial q^i}.$$

This is not sufficient for flatness of y^1, \dots, y^p since it is possible that y^1, \dots, y^p are differentially dependent and this happens when E does not depend on q^n . More precisely y^1, \dots, y^p are differentially dependent around q when there exists a neighbourhood V of q such that $\frac{\partial E}{\partial q^n}$ is identically zero on $(\pi_2^{-1}(V) \cap \{E = 0\}) \subset J^2(\mathbb{R}, Q)$ where $\pi_2 : J^2(\mathbb{R}, Q) \rightarrow Q$ is the standard projection. The functions E and $\frac{\partial E}{\partial q^n}$ are both affine in \ddot{y} and quadratic in \dot{y} with the coefficients functions only of (y, q^n) and E depends on \ddot{y} non trivially since metric g is non degenerate. Hence $\frac{\partial E}{\partial q^n}$ is either identically zero on $\pi_2^{-1}(q) \cap \{E = 0\}$ or it is non zero for generic points on $\pi_2^{-1}(q) \cap \{E = 0\}$. Further more $\frac{\partial E}{\partial q^n}$ is identically zero on $\pi_2^{-1}(q) \cap \{E = 0\}$ if and only if it is a multiple of E as a polynomial in \dot{y} and \ddot{y} for points on $\pi_2^{-1}(q)$. Hence the regularity condition we impose is that $\frac{\partial E}{\partial q^n}$ is a not a multiple of E as a polynomial in \dot{y} and \ddot{y} for points on $\pi_2^{-1}(q)$. Then it would follow from continuity and implicit function theorem that for generic points on $\pi_2^{-1}(V) \cap \{E = 0\}$ where V is some neighbourhood of q , q^n can be locally solved for in terms of y, \dot{y}, \ddot{y} , implying flatness around q .

What is left to be shown is that this condition translates to the regularity condition stated in the proposition. It is sufficient to show that $\frac{\partial E}{\partial q^n}$ is a multiple of E as polynomials in \dot{y}, \ddot{y} with the ratio being a smooth function on Q is equivalent to the set of ratios of functions (15) all being identically equal in a neighbourhood of q . The rest of the proof is basically algebra and we refer the reader to [13].

4. Example: Underwater Vehicle

We shall study a simple model of underwater vehicle that is controlled by a force applied through a fixed

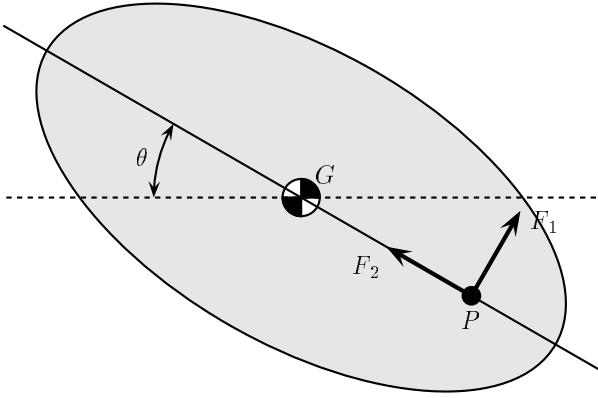


Fig. 1: Underwater vehicle in \mathbb{R}^2

point P on the body whose magnitude and direction can be independently controlled.

Only the motion in the vertical plane is considered and hence our configuration space is $SE(2) = \mathbb{R}^2 \times S^1$. This is reasonable when the vehicle has symmetries about 3 orthogonal planes. The kinetic energy is given by

$$\begin{aligned} & \frac{1}{2}(m + \delta m)(\dot{x}_1 \cos \theta - \dot{x}_2 \sin \theta)^2 \\ & + \frac{1}{2}(m - \delta m)(\dot{x}_1 \sin \theta + \dot{x}_2 \cos \theta)^2 + \frac{1}{2}I(\dot{\theta})^2, \quad (21) \end{aligned}$$

where (x_1, x_2) are horizontal and vertical coordinates of the centre of mass G , θ is the orientation of line PG with respect to horizontal axis, $m = M + (m_1 + m_2)/2$ and $\delta m = (m_1 - m_2)/2$ where M is the mass of the vehicle and m_1 and m_2 are added mass terms that take into account inertia of the fluid, and I is the effective moment of inertia taking into account the fluid. This model neglects viscosity effects, but takes into account approximately the effect of the momentum of the displaced water (see [14]). If the vehicle is in air (strictly speaking vacuum) $m_1 = m_2 = 0$, so $m = M$ and $\delta m = 0$ and the kinetic energy takes the familiar form

$$\frac{1}{2}(m(\dot{x}_1)^2 + m(\dot{x}_2)^2 + I(\dot{\theta})^2)$$

where I is the usual moment of inertia and the model is the same as that of VTOL (see [15]).

The metric g in coordinates x_1, x_2, θ is given by the matrix

$$\begin{bmatrix} m + \delta m \cos 2\theta & -\delta m \sin 2\theta & 0 \\ -\delta m \sin 2\theta & m - \delta m \cos 2\theta & 0 \\ 0 & 0 & I \end{bmatrix}.$$

The control forces lie in the codistribution

$$\begin{aligned} P &= \text{span}\{d(x_1 + R \cos \theta), d(x_2 - R \sin \theta)\} \\ &= \text{span}\{dx_1 - R \sin \theta d\theta, dx_2 - R \cos \theta d\theta\} \end{aligned}$$

and $\xi = \frac{\partial}{\partial \theta} + R \cos \theta \frac{\partial}{\partial x_1} + R \sin \theta \frac{\partial}{\partial x_2}$ spans $\text{ann } P$.

The Christoffel symbols Γ_{jk}^i can be computed from g using equation (10). Then using formula (11) we see

that

$$\begin{aligned} \nabla_{\frac{\partial}{\partial x_1}} \xi &= -\frac{m\delta m}{m^2 - (\delta m)^2} \sin 2\theta \frac{\partial}{\partial x_1} \\ & - \frac{\delta m}{m^2 - (\delta m)^2} (\delta m + m \cos 2\theta) \frac{\partial}{\partial x_2} + \frac{R\delta m \sin 3\theta}{I} \frac{\partial}{\partial \theta} \\ \nabla_{\frac{\partial}{\partial x_2}} \xi &= -\frac{\delta m}{m^2 - (\delta m)^2} (-\delta m + m \cos 2\theta) \frac{\partial}{\partial x_1} \\ & - \frac{m\delta m}{m^2 - (\delta m)^2} \sin 2\theta \frac{\partial}{\partial x_2} + \frac{R\delta m \cos 3\theta}{I} \frac{\partial}{\partial \theta} \\ \nabla_{\frac{\partial}{\partial \theta}} \xi &= -\frac{R}{m^2 - (\delta m)^2} (m\delta m \sin 3\theta \\ & + m^2 \sin \theta - 2(\delta m)^2 \sin \theta) \frac{\partial}{\partial x_1} \\ & + \frac{R}{m^2 - (\delta m)^2} (-m\delta m \cos 3\theta \\ & + m^2 \cos \theta - 2 \cos \theta (\delta m)^2) \frac{\partial}{\partial x_2} \quad (22) \end{aligned}$$

It can be seen by computing the appropriate determinant that above vector fields are linearly independent for generic points and generic parameter values $m, \delta m, I, R$. By equation (13)

$$D = \text{span}\{\nabla_{\frac{\partial}{\partial x_1}} \xi, \nabla_{\frac{\partial}{\partial x_2}} \xi, \nabla_{\frac{\partial}{\partial \theta}} \xi, \xi\}.$$

$D = TQ$ for generic points on Q , for generic parameter values and hence the system is not configuration flat for generic parameter values regardless of the potential energy function.

However for the case $\delta m = 0$, corresponding to vehicle in air, we see that

$$\begin{aligned} D &= \text{span}\left\{-mR \sin \theta \frac{\partial}{\partial x_1} + mR \cos \theta \frac{\partial}{\partial x_2}, \right. \\ & \left. mR \cos \theta \frac{\partial}{\partial x_1} + mR \sin \theta \frac{\partial}{\partial x_2} + I \frac{\partial}{\partial \theta}\right\}. \end{aligned}$$

Hence $\dim D = 2$ and $\eta = \frac{\partial}{\partial \theta} - \frac{I}{mR} \cos \theta \frac{\partial}{\partial x_1} - \frac{I}{mR} \sin \theta \frac{\partial}{\partial x_2}$ spans the orthogonal complement to D . Since D has codimension 1, up to a diffeomorphism there is at most 1 set of flat outputs. One set of functions that ‘‘cut out’’ the foliation due to η is

$$y_1 = x_1 - \frac{I}{mR} \cos \theta, \quad y_2 = x_2 + \frac{I}{mR} \sin \theta.$$

To ensure that y_1, y_2 are indeed flat outputs we must check the regularity conditions (16). Let us choose $z = \theta$ as a complementary coordinate to y_1, y_2 . Then,

$$\begin{aligned} \frac{\partial}{\partial y_1} &= \frac{\partial}{\partial x_1}, \quad \frac{\partial}{\partial y_2} = \frac{\partial}{\partial x_2}, \\ \frac{\partial}{\partial z} &= -\frac{I}{mR} \sin \theta \frac{\partial}{\partial x_1} - \frac{I}{mR} \sin \theta \frac{\partial}{\partial x_2} + \frac{\partial}{\partial \theta}. \end{aligned}$$

Hence

$$\begin{aligned} \frac{\partial}{\partial z} \left(g(\xi, \frac{\partial}{\partial y_1}) \right) : g(\xi, \frac{\partial}{\partial y_1}) &= -\sin z : \cos z \\ \frac{\partial}{\partial z} \left(g(\xi, \frac{\partial}{\partial y_2}) \right) : g(\xi, \frac{\partial}{\partial y_2}) &= \cos z : \sin z. \quad (23) \end{aligned}$$

So at any point $q = (y_1, y_2, z)$ these two ratios are unequal. This ensures that y_1, y_2 are indeed flat outputs everywhere. In fact the $\delta m = 0$ case is already known to be flat (see [15, 8]). We have just shown that up to a diffeomorphism these are the only configuration flat outputs. This result is independent of any assumptions we make on the potential function V . Now let us suppose the system is moving under gravity and the potential energy is given by $V = mgx_2$ where $g \approx 9.8$ m/s² is the acceleration due to gravity. Then the solutions of the system in coordinates y_1, y_2, z satisfy the ODE

$$\ddot{y}_1 \sin z + \ddot{y}_2 \cos z + g \cos z = 0.$$

So along generic solution curves we get,

$$z(t) = \tan^{-1} \frac{\ddot{y}_2 + g}{\ddot{y}_1}$$

or

$$z(t) = \tan^{-1} \frac{\ddot{y}_2 + g}{\ddot{y}_1} + \pi.$$

The exception being the singularity at $\ddot{y}_1 = 0, \ddot{y}_2 + g = 0$. Note that this singularity is not a point on Q but corresponds to a submanifold in the jet space $J^2(\mathbb{R}, Q)$, the space with coordinates $(t, q, \dot{q}, \ddot{q})$ and such singularities are very common in practical examples. We still want to regard such systems as flat and this is the reason why our definition of flatness refers to generic curves as opposed to all curves. Also note that though potential V does not affect the flat outputs of the system it influences where the singularities occur.

5. Conclusions and future work

We have presented a method for determining configuration flatness of Lagrangian control systems with n degrees of freedom and $n - 1$ controls. Our method is constructive and provides a way for finding configuration flat outputs if they exist. We assumed a Lagrangian of the form “kinetic energy minus potential”. We also assumed that the range of control forces only depends on configuration. These assumptions are not unreasonable since a wide class of systems fall into this category. However $n - 1$ controls is a special case and is the simplest case next to fully actuated (n controls) systems which are allways flat. In that sense we regard this as a first step towards a general theory of configuration flatness of Lagrangian systems. The authors are currently working on generalising this result to arbitrary number of controls.

Acknowledgement It is a pleasure to thank Jerrold Marsden for valuable comments.

References

[1] M. Fliess, J. Lévine, P. Martin, and P. Rouchon. Flatness and defect of nonlinear systems: Introductory theory and examples. *International Journal of Control*, 61(6):1327–1361, June 1995.

[2] M. Fliess, J. Levine, P. Martin, and P. Rouchon. Linéarisation par bouclage dynamique et transformations de Lie-Bäcklund. *Comptes Rendus des Séances de l’Académie des Sciences*, 317(10):981–986, November 1993. Serie I.

[3] P. Martin. *Contribution à l’étude des systèmes différentiellement plats*. PhD thesis, École des Mines de Paris, 1992.

[4] J.-B. Pomet. A differential geometric setting for dynamic equivalence and dynamic linearization. To appear in *Proceedings Banach Center Publications*, 94 or 95. Also available as technical report INRIA No 2312, Oct 93.

[5] M. van Nieuwstadt, M. Rathinam, and R.M. Murray. Differential flatness and absolute equivalence. In *Proceedings IEEE Control and Decision Conference*, pages 326–332, December 1994.

[6] M. Rathinam and W.M. Sluis. A test for differential flatness by reduction to single input systems. In *Proceedings of IFAC 96*, volume E, pages 257–262. Also available as Technical Memorandum CIT-CDS 95-018, California Institute of Technology, Pasadena, CA 91125, June 95.

[7] M. van Nieuwstadt and R.M. Murray. Approximate trajectory generation for differentially flat systems with zero dynamics. In *Proceedings IEEE Control and Decision Conference*, pages 4224–4230, December 1995.

[8] R.M. Murray, M. Rathinam, and W.M. Sluis. Differential flatness of mechanical control systems. 1995. *Proceedings of the 1995 ASME International Congress and Exposition*.

[9] W.F. Shadwick. Absolute equivalence and dynamic feedback linearization. *Systems & Control Letters*, 15:35–39, 1990.

[10] W.M. Sluis. *Absolute Equivalence and its Applications to Control Theory*. PhD thesis, University of Waterloo, 1993.

[11] D.J. Saunders. *The Geometry of Jet Bundles*. Cambridge University Press, 1989.

[12] R. Abraham and J.E. Marsden. *Foundations of Mechanics*. Addison Wesley, 2nd edition, 1985.

[13] M. Rathinam and R.M. Murray. Configuration flatness of lagrangian systems underactuated by one control. Technical Memorandum CIT-CDS 96-006, California Institute of Technology, Pasadena, CA 91125, March 96. Also submitted to SIAM journ. on Control and Optimization.

[14] N.E. Leonard. Compensation for actuator failures: Dynamics and control of underactuated underwater vehicles. 1995. *Proceedings of the 9th International Symposium on Unmanned Untethered Submersible Technology*.

[15] Ph. Martin, S.Devasia, and B. Paden. A different look at output tracking: control of a vtol aircraft. *Automatica*, 32(1):101–107, January 1996.