



## CDS 101: Lecture 8.1 Frequency Domain Design



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19 November 2002

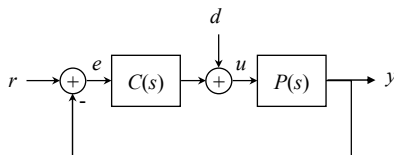
### Goals:

- Describe the use of frequency domain performance specification
- Show how to use “loop shaping” to achieve a performance specification
- Work through a detailed example of a control design problem

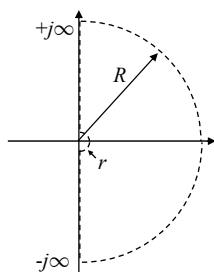
### Reading:

- No new reading this week
- *Advanced*: Lewis, Chapter 12

## Review from Last Week



- Nyquist criteria for loop stability
- Gain, phase margin for robustness



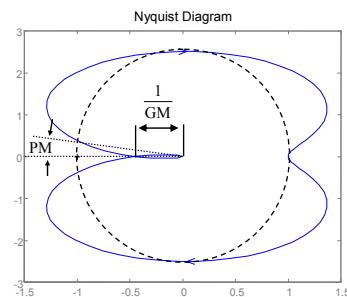
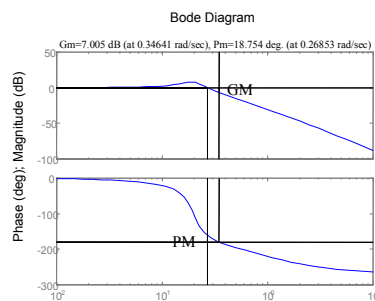
**Thm (Nyquist).**

$P$  # RHP poles of  $L(s)$

$N$  # CW encirclements

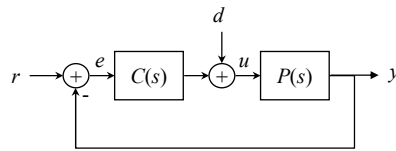
$Z$  # RHP zeros

$$Z = N + P$$



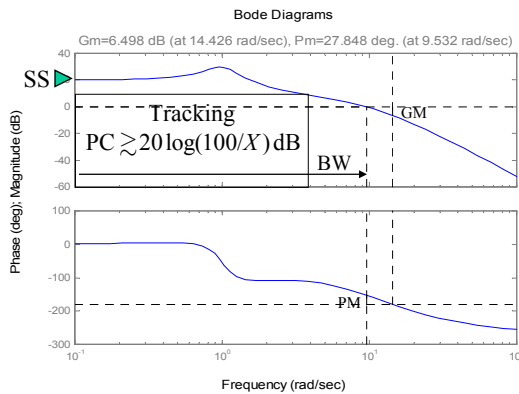
## Frequency Domain Performance Specifications

Specify bounds on the loop transfer function to guarantee desired performance



$$L(s) = P(s)C(s)$$

$$H_{er} = \frac{1}{1+L} \quad H_{yr} = \frac{L}{1+L}$$



- Steady state error:

$$H_{er}(0) = 1/(1+L(0)) \approx 1/L(0)$$

$\Rightarrow$  zero frequency ("DC") gain  $\rightarrow$

- Bandwidth: assuming  $\sim 90^\circ$  phase margin

$$\frac{L}{1+L}(j\omega_c) \approx \left| \frac{1}{1+j} \right| = \frac{1}{\sqrt{2}}$$

$\Rightarrow$  sets crossover freq  $\rightarrow$

- Tracking:  $X\%$  error up to frequency  $\omega_l \Rightarrow$  determines gain bound ( $1+PC > 100/X$ )  $\square$

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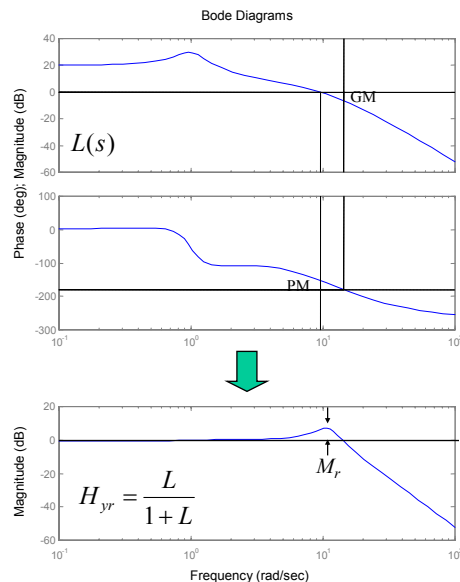
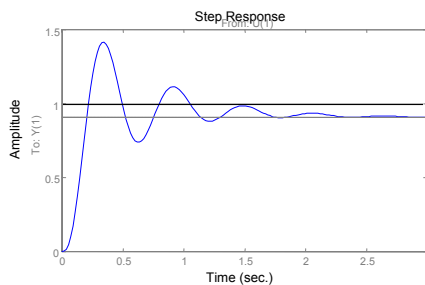
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## Relative Stability

Relative stability: how stable is system to disturbances at certain frequencies?

- System can be stable but still have very bad response at certain frequencies
- Typically occurs if system has low phase margin  $\Rightarrow$  get resonant peak in closed loop ( $M_r$ ) + poor step response
- Solution: specify minimum phase margin. Typically  $45^\circ$  or more

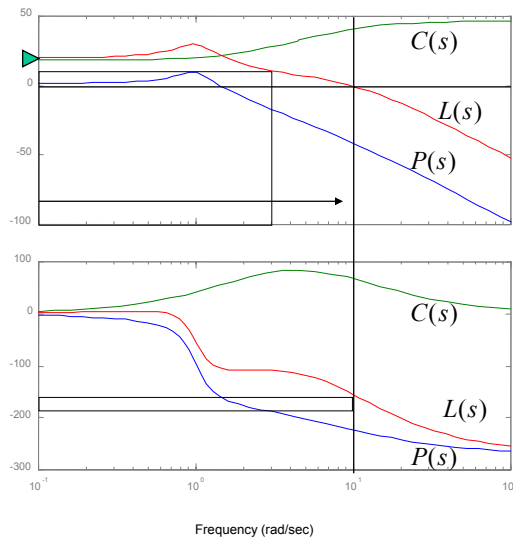


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## Overview of Loop Shaping



### Performance specification

- ▶ Steady state error
- Tracking error
- Bandwidth
- Relative stability

### Approach: “shape” loop transfer function using $C(s)$

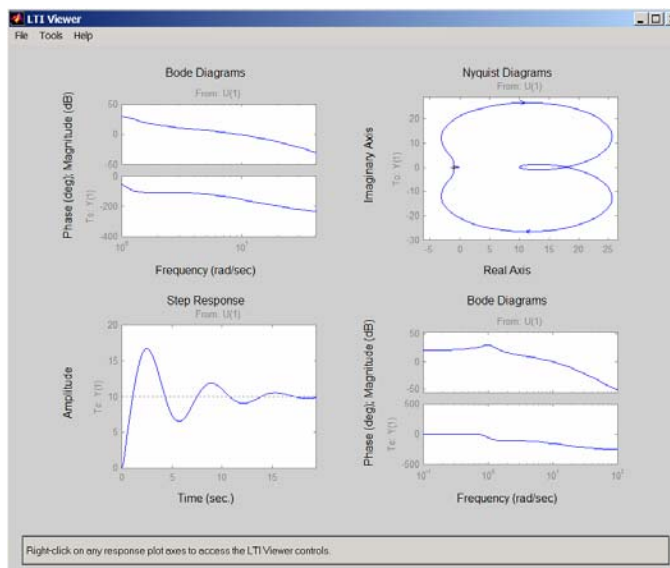
- $P(s)$  + specifications given
- $L(s) = P(s) C(s)$ 
  - Use  $C(s)$  to choose desired shape for  $L(s)$
- Important: can't set gain and phase independently

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## LTIVIEW



### MATLAB LTIVIEW

- Allows simultaneous view of up to six plots
- Can apply to any system in the MATLAB workspace
- Useful for seeing how the various concepts relate to each others

### Caution

- Doesn't allow much control over details of each plot

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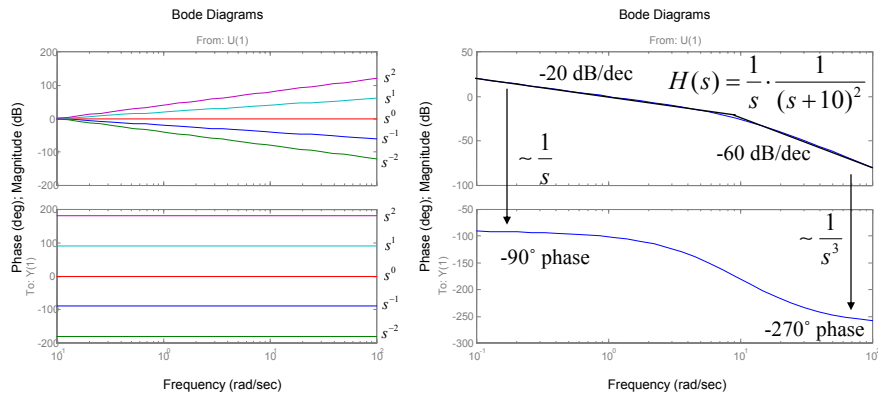
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## Gain/phase relationships

**Gain and phase for transfer function with real coefficients are not independent**

- Given a given shape for the gain, there is a unique “minimum phase” transfer function that achieves that gain at the specified frequencies
- Basic idea: slope of the gain determines the phase
- Implication: you have to tradeoff gain versus phase in control design



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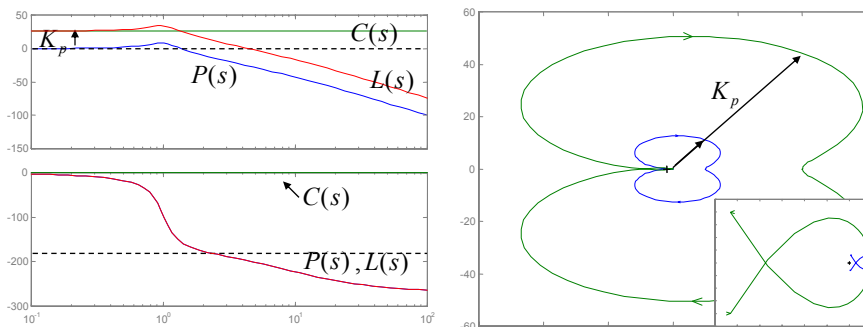
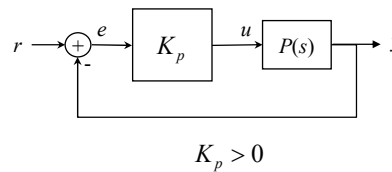
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## Proportional Feedback

**Simplest controller choice:  $u = K_p e$**

- Effect: lifts gain with no change in phase
- Good for plants with low phase up to desired bandwidth
- Bode: shift gain up by factor of  $K_p$
- Nyquist: scale Nyquist contour



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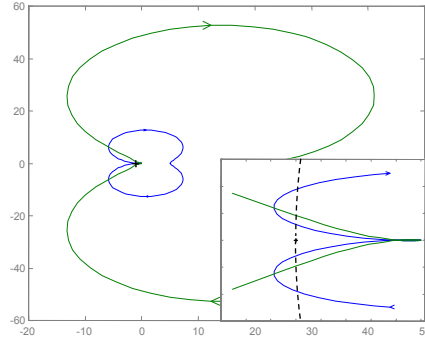
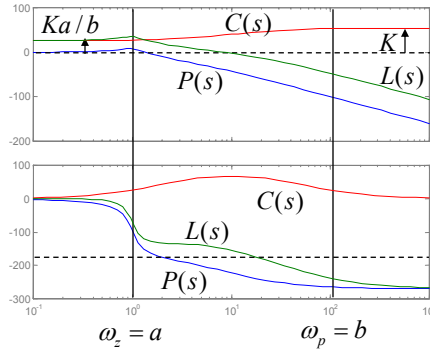
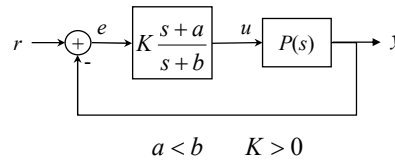
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## Lead compensation

### Use to increase phase in frequency band

- Effect: lifts phase by increasing gain at high frequency
- *Very* useful controller; increases PM
- Bode: add phase between zero and pole
- Nyquist: increase phase margin



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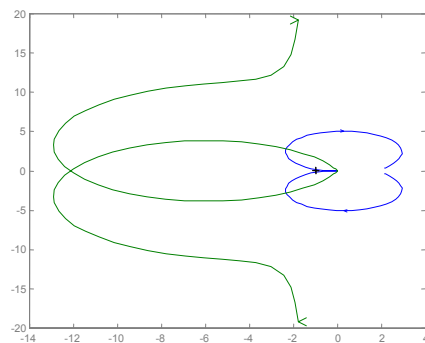
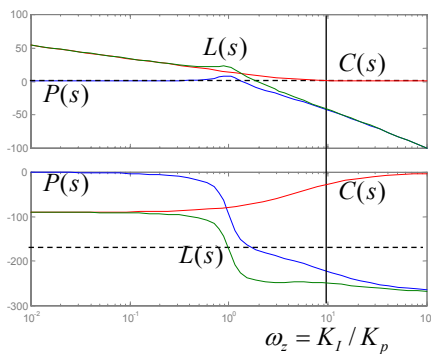
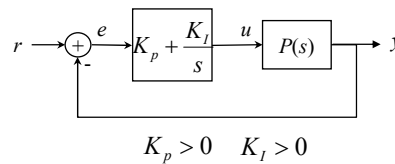
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## Proportional + Integral Compensation

### Use to eliminate steady state error

- Effect: lifts gain at low frequency
- Gives *zero* steady state error
- Bode: shift gain up by factor of  $K_p$
- Nyquist: scale Nyquist contour



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### Example: Pitch Control for Caltech Ducted Fan



#### System description

- Vector thrust engine attached to wing
- Inputs: fan thrust, thrust angle (vectored)
- Outputs: position and orientation
- States:  $x, y, \theta$  + derivatives
- Dynamics: flight aerodynamics

#### Control approach

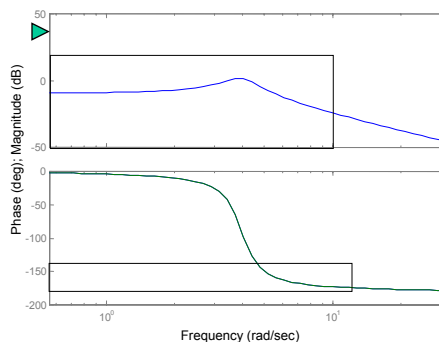
- Design “inner loop” control law to regulate pitch ( $\theta$ ) using thrust vectoring
- Second “outer loop” controller regulates the position and altitude by commanding the pitch and thrust
- Basically the same approach as aircraft control laws

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### Performance Specification and Design Approach



#### Performance Specification

- $\leq 1\%$  steady state error
  - Zero frequency gain  $> 100$
- $\leq 10\%$  tracking error up to 10 rad/sec
  - Gain  $> 10$  from 0-10 rad/sec
- $\geq 45^\circ$  phase margin
  - Gives good relative stability
  - Provides robustness to uncertainty

$$P(s) = \frac{r}{Js^2 + ds + mgl}$$

$$C(s) = K \frac{s+a}{s+b}$$

$$a = 25$$

$$b = 300$$

$$K = 15 \cdot 300$$

#### Design approach

- Open loop plant has poor phase margin
- Add phase lead in 5-50 rad/sec range
- Increase the gain to achieve steady state and tracking performance specs
- Avoid integrator to minimize phase

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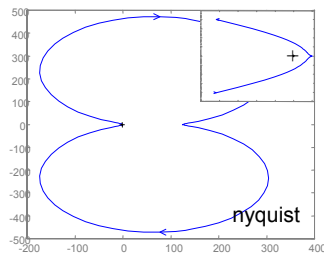
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## Control Design and Analysis

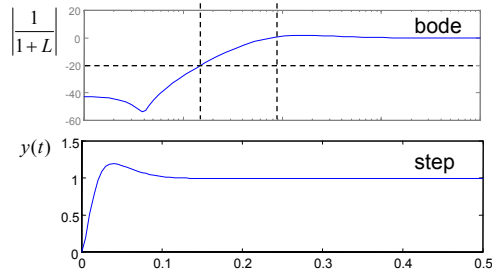
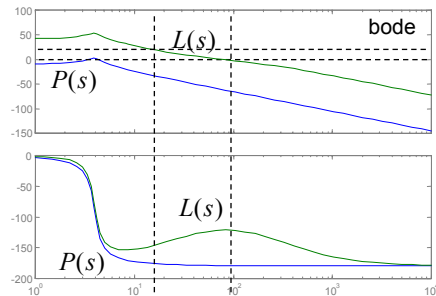
### Select parameters to satisfy specs

- Place phase lead in desired crossover region (given by desired BW)
- Phase lead peaks at 10X of zero location
- Place pole sufficiently far out to insure that phase does not decrease too soon
- Set gain as needed for tracking + BW
- Verify controller using Nyquist plot, etc



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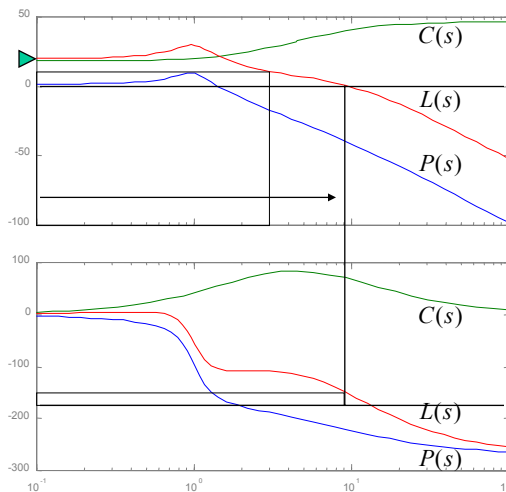


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## Summary: Frequency Domain Design

### Loop Shaping for Stability and Performance

- Steady state error, bandwidth, tracking



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### Main ideas

- Performance specifications give bounds on loop transfer function
- Use controller to shape response
- Gain/phase relationships constrain design approach
- Standard compensators: proportional, lead, PI



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