

Invariant Manifolds, 3-Body Problem & Petit Grand Tour of Jovian Moons

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■ Acknowledgements

- ▶ H. Poincaré, J. Moser
- ▶ C. Conley, R. McGehee, D. Appleyard
- ▶ C. Simó, J. Llibre, R. Martinez
- ▶ B. Farquhar, D. Dunham
- ▶ E. Belbruno, B. Marsden, J. Miller
- ▶ K. Howell, B. Barden, R. Wilson

■ Outline of Presentation

► Main Theme

- how to use invariant manifold of 3-body problem in space mission design.

► Background and Motivation

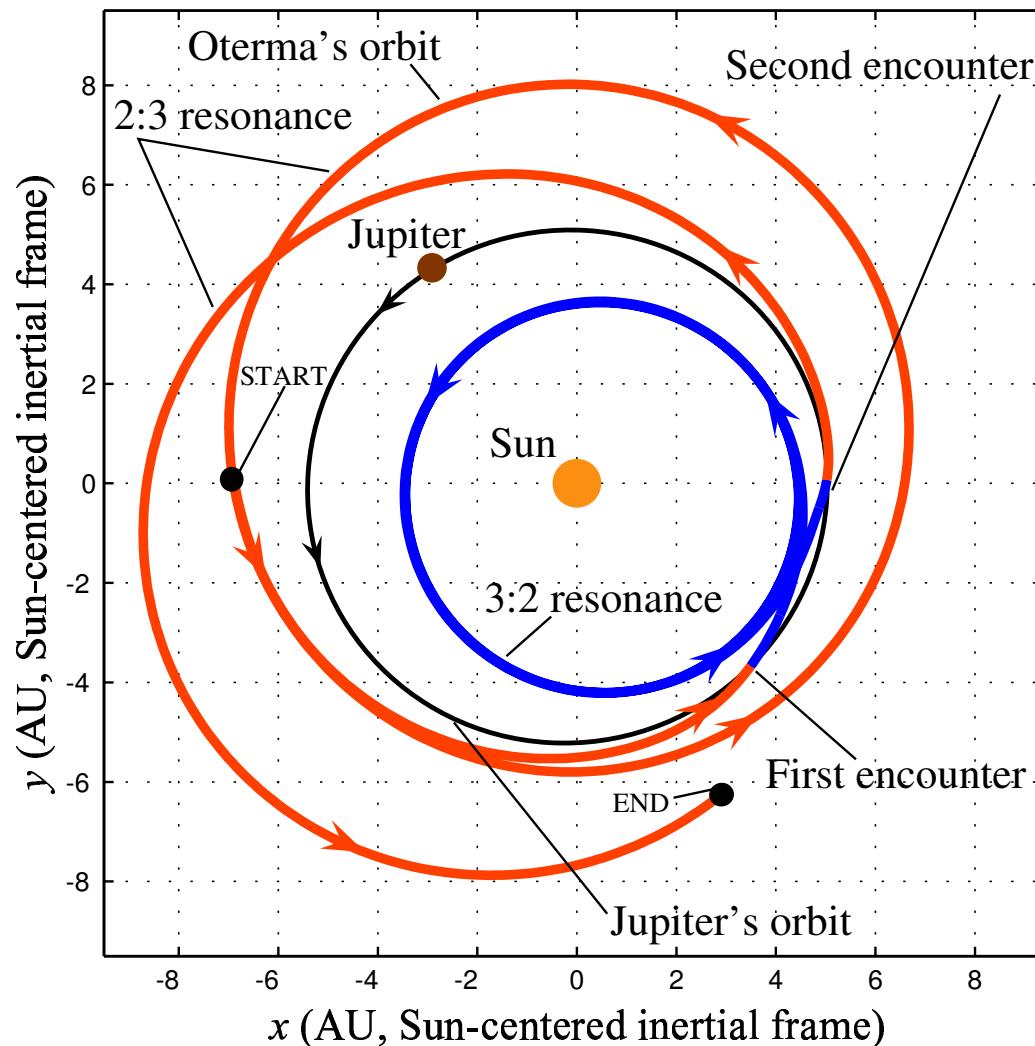
► Major Results and Some Technical Details.

► Low Energy Transfer between Jovian Moons.

► Conclusion and Ongoing Work.

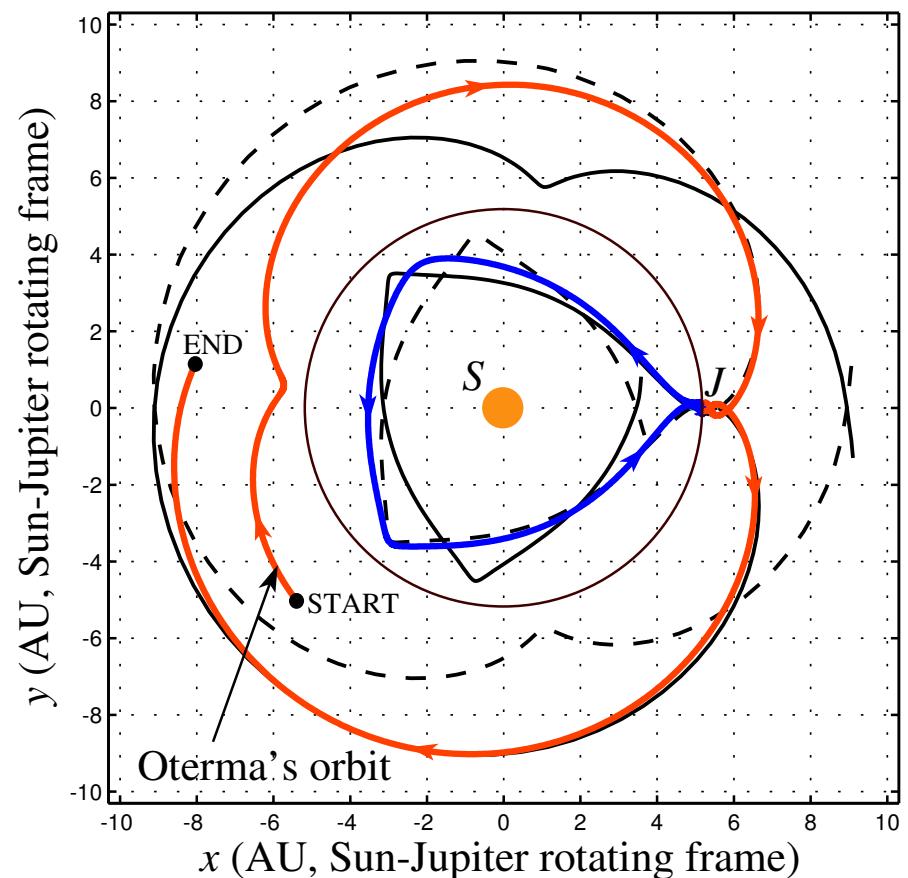
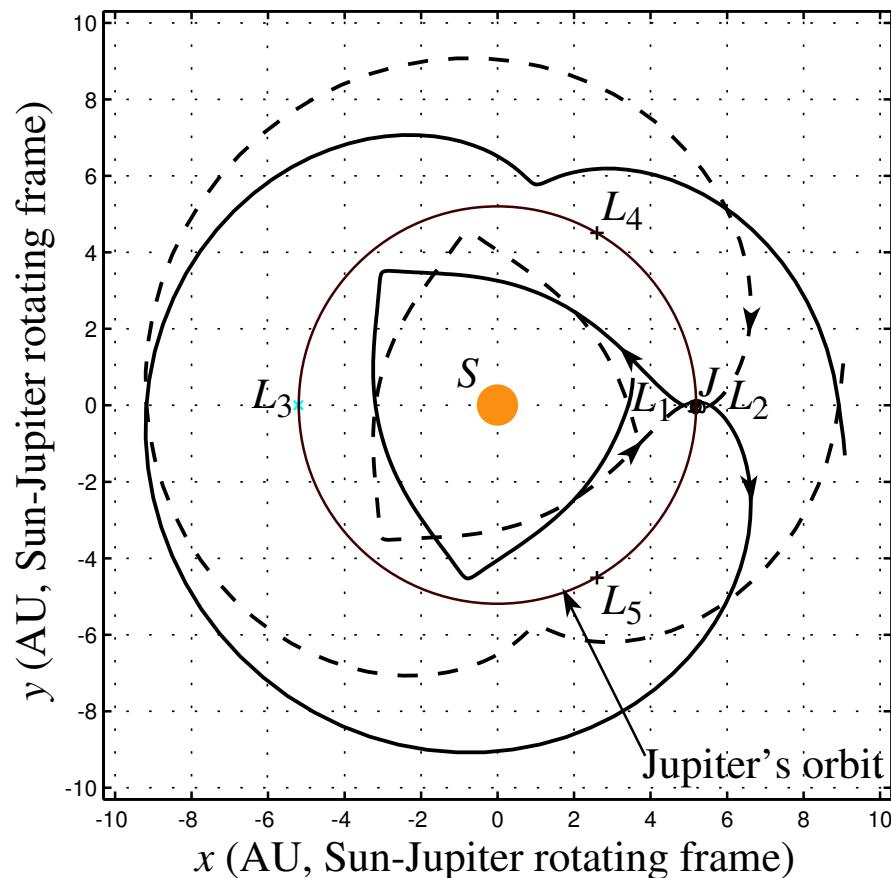
Jupiter Comets

- Rapid transition from **outside** to **inside** Jupiter's orbit.
- Captured temporarily by Jupiter during transition.
- **Exterior** (2:3 resonance). **Interior** (3:2 resonance).



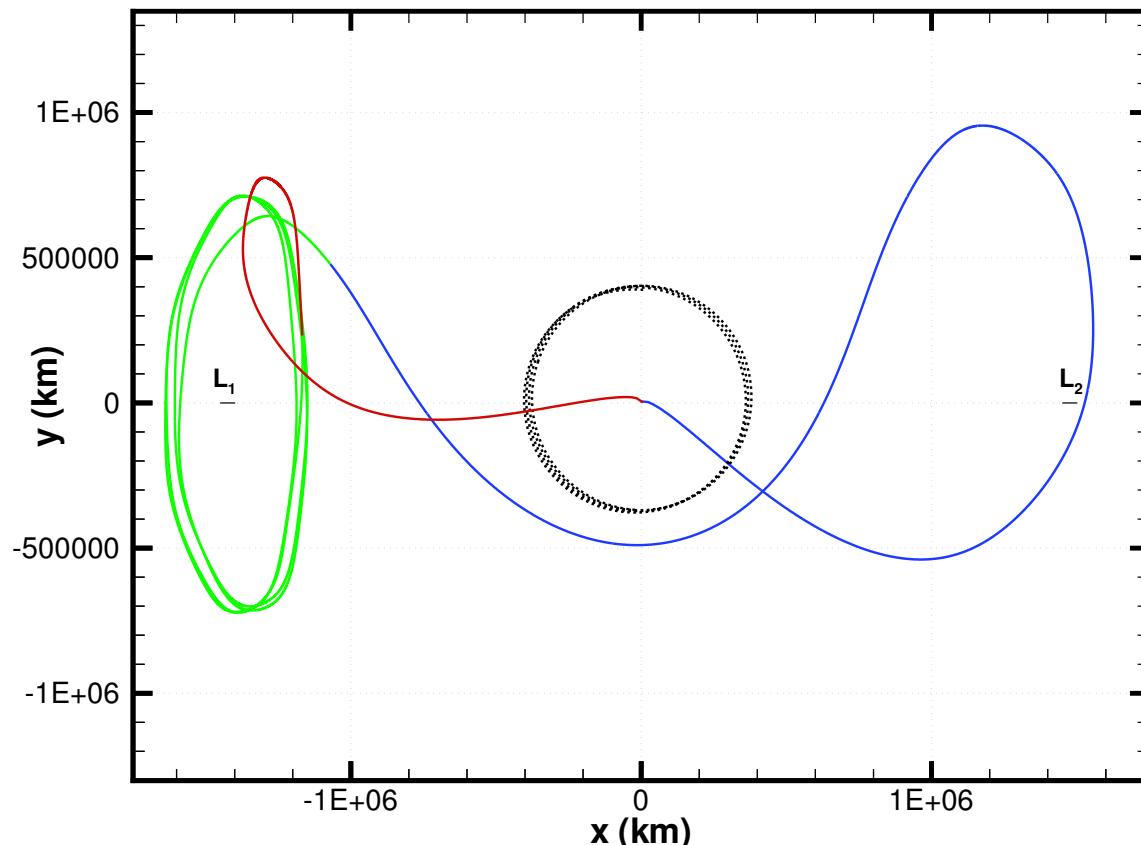
■ Jupiter Comets

- ▶ Belbruno and B. Marsden [1997]
- ▶ Lo and Ross [1997]
 - Comet in **rotating frame** follows **invariant manifolds**.
- ▶ Moser, Conley and McGehee. LLibre, Martinez and Simó [1985].



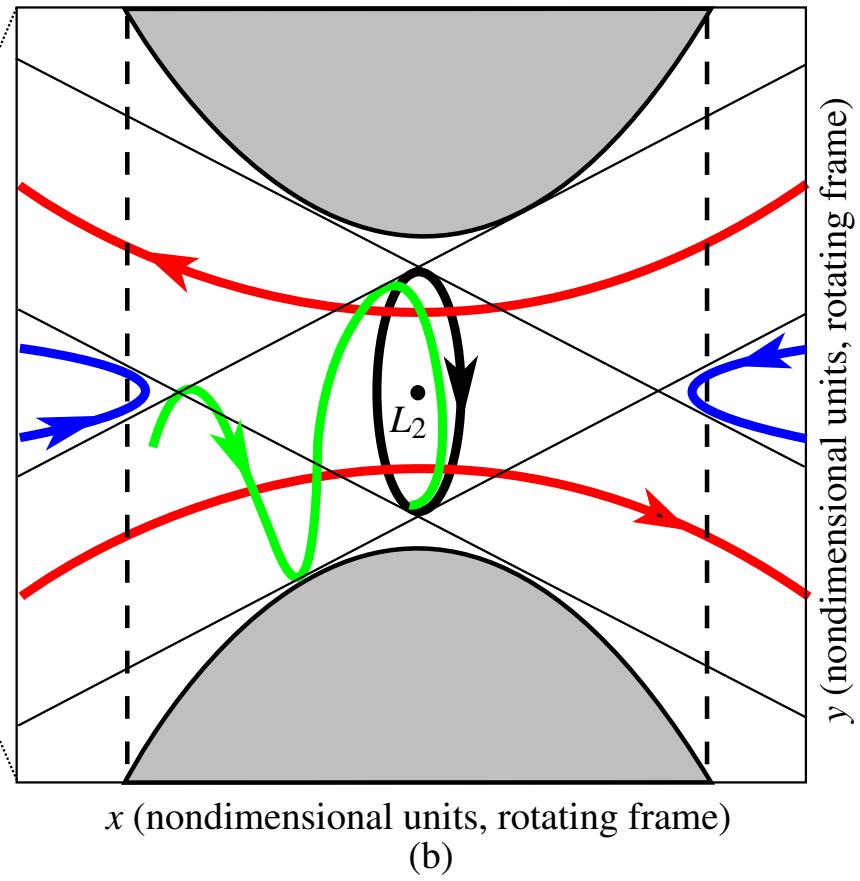
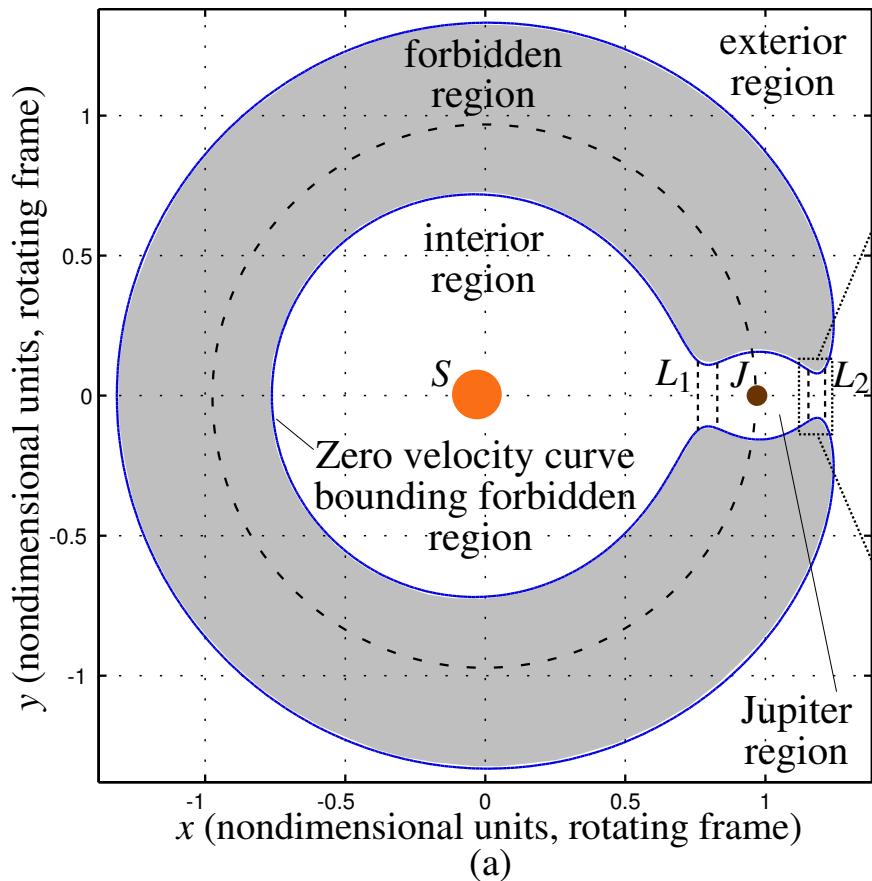
■ Genesis Discovery Mission

- ▶ To understand better **heteroclinic dynamics** used by **Genesis Mission** in returning solar wind sample to Earth.
 - Howell, Lo, Barden and Wilson.
- ▶ **Halo orbit, transfer/ return** trajectories in rotating frame.



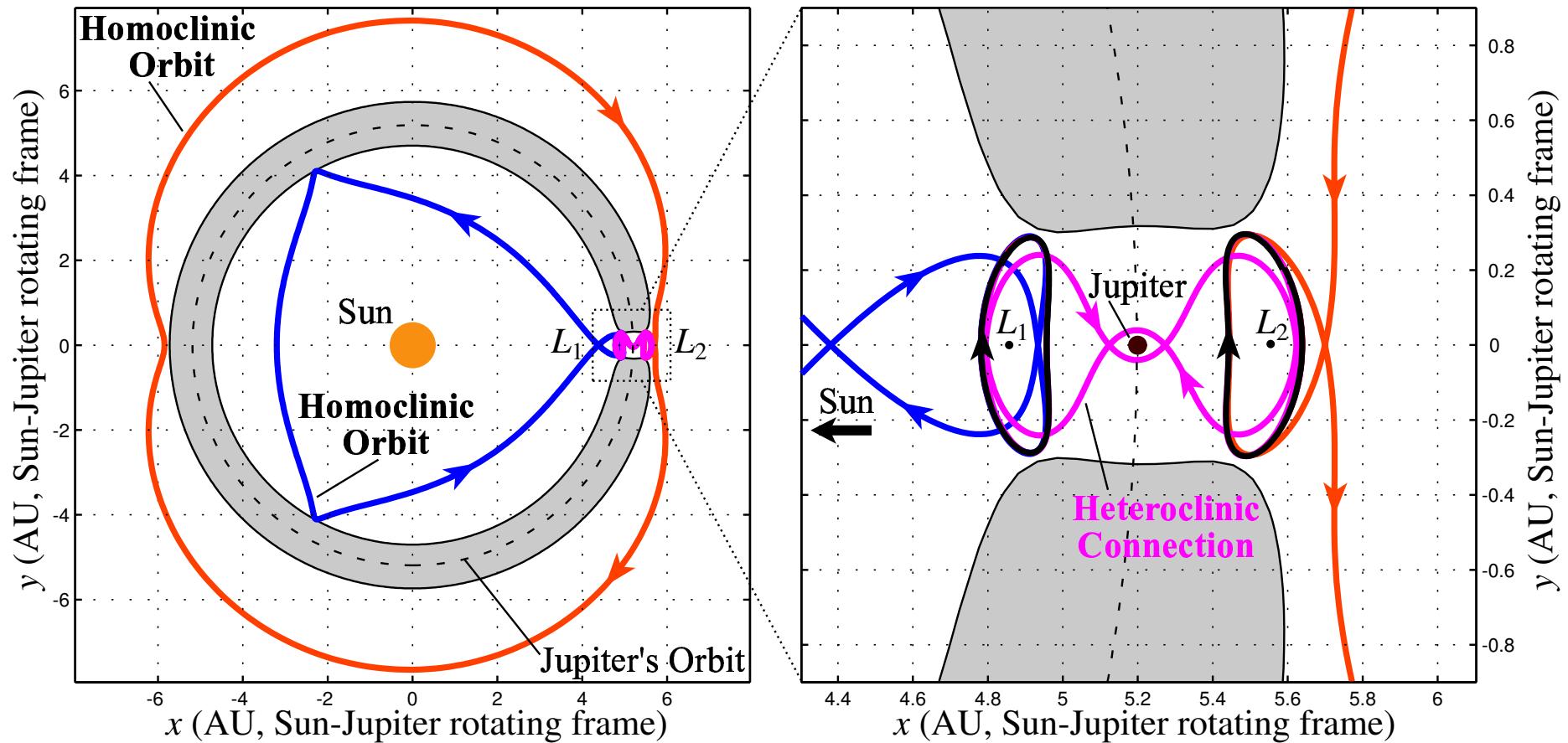
The Flow near L_1 and L_2

- For energy value just above that of L_2 ,
Hill's region contains a “neck” about L_1 & L_2 .
- Comet can make **transition** through these equilibrium regions.
- 4 types of orbits:
periodic, **asymptotic**, **transit** & **nontransit**.



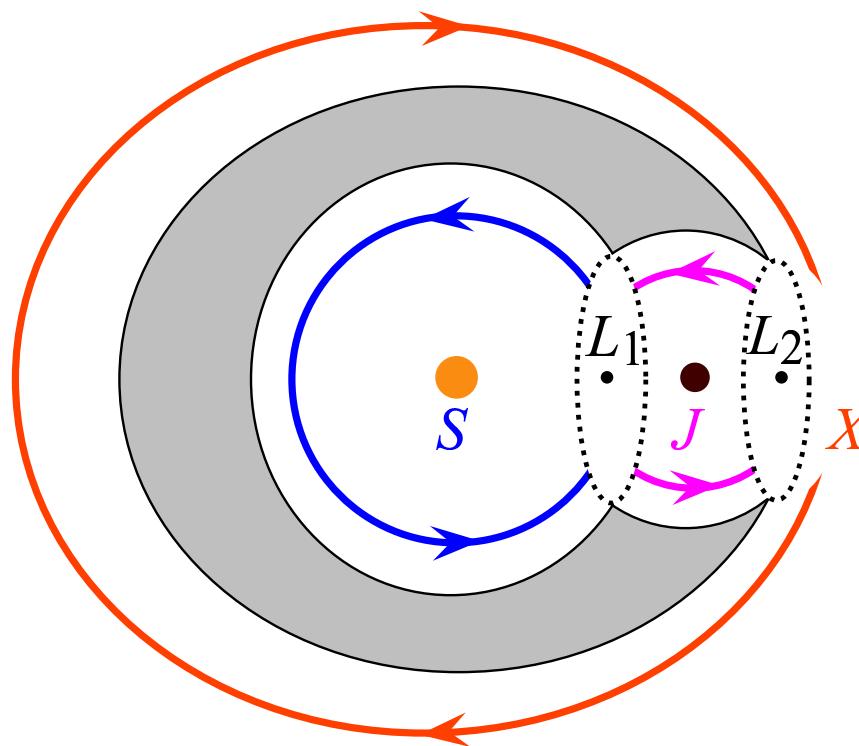
■ Major Result (A): Heteroclinic Connection

- Found **heteroclinic connection** between pair of periodic orbits.
- Found a large class of **orbits** near this (homo/heteroclinic) **chain**.
- Comet can follow these **channels** in rapid transition.



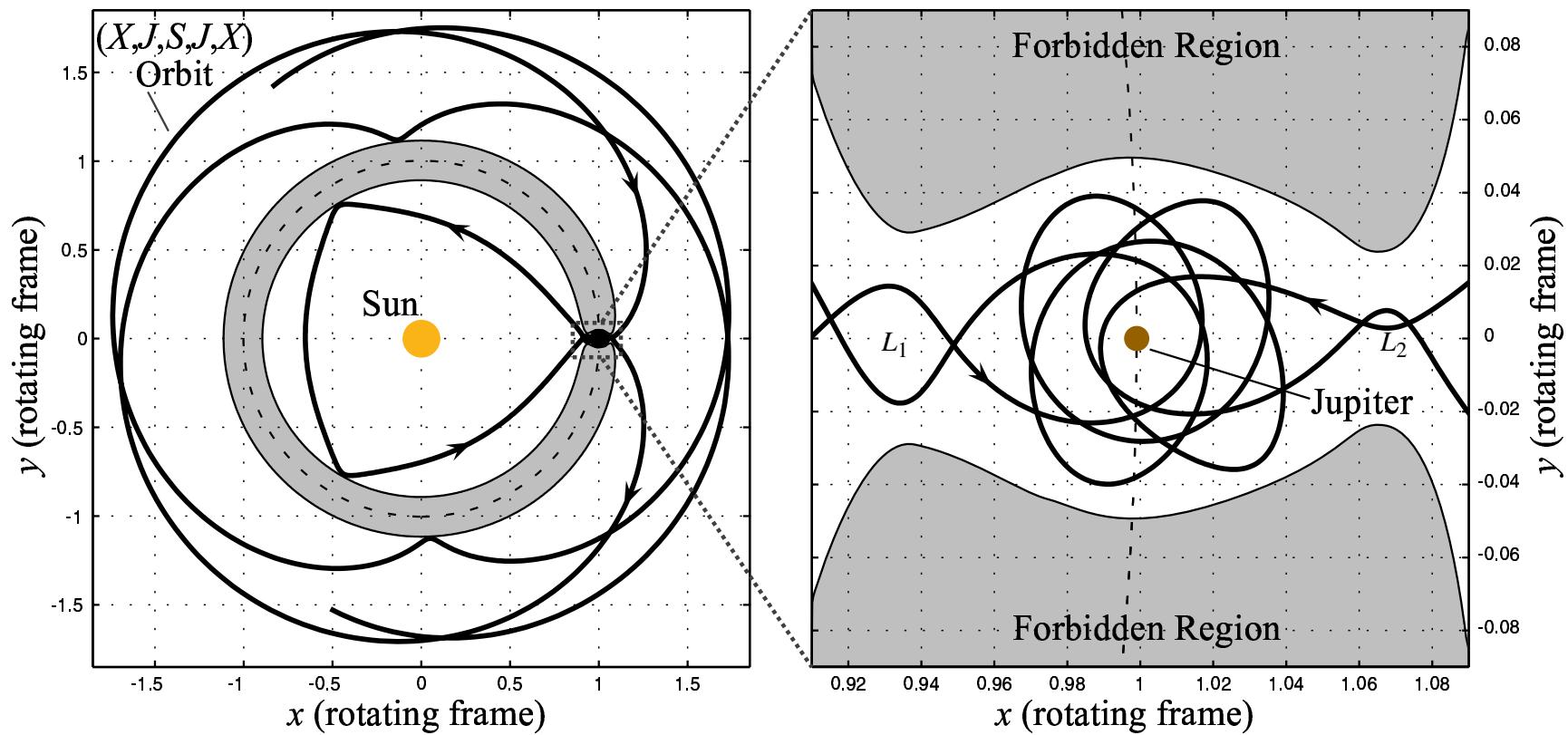
■ Major Result (B): Existence of Transitional Orbit

- ▶ **Symbolic sequence** used to label itinerary of each comet orbit.
- ▶ **Main Theorem:** For any admissible **itinerary**,
e.g., $(\dots, \mathbf{X}, \mathbf{J}; \mathbf{S}, \mathbf{J}, \mathbf{X}, \dots)$, there exists an orbit whose
whereabouts matches this **itinerary**.
- ▶ Can even specify **number of revolutions** the comet makes
around Sun & Jupiter (plus L_1 & L_2).



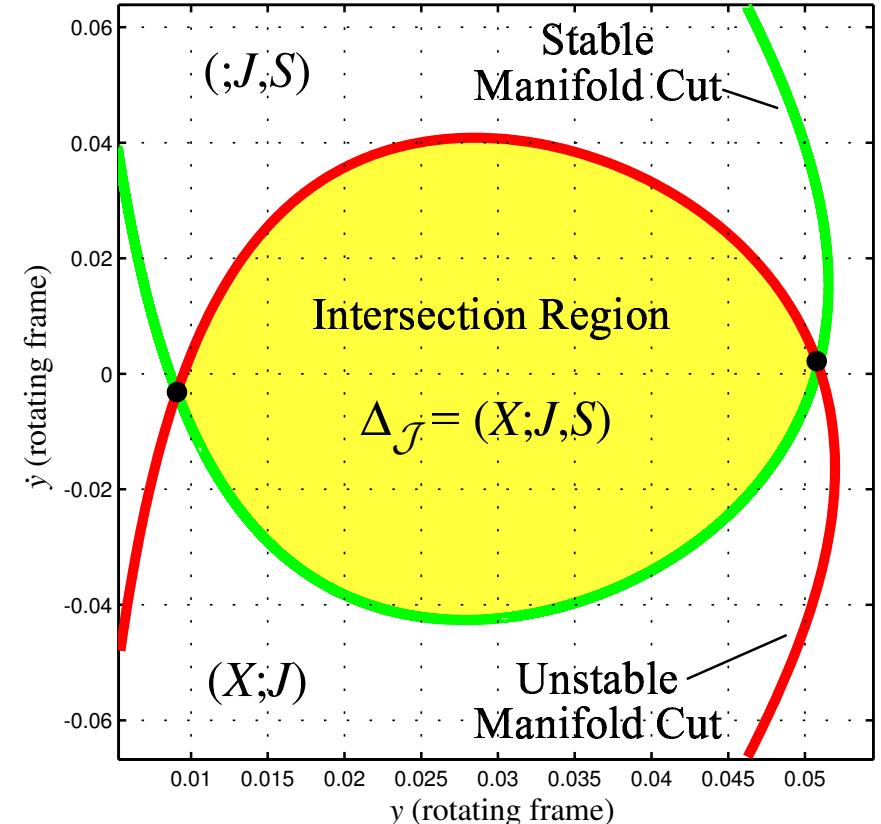
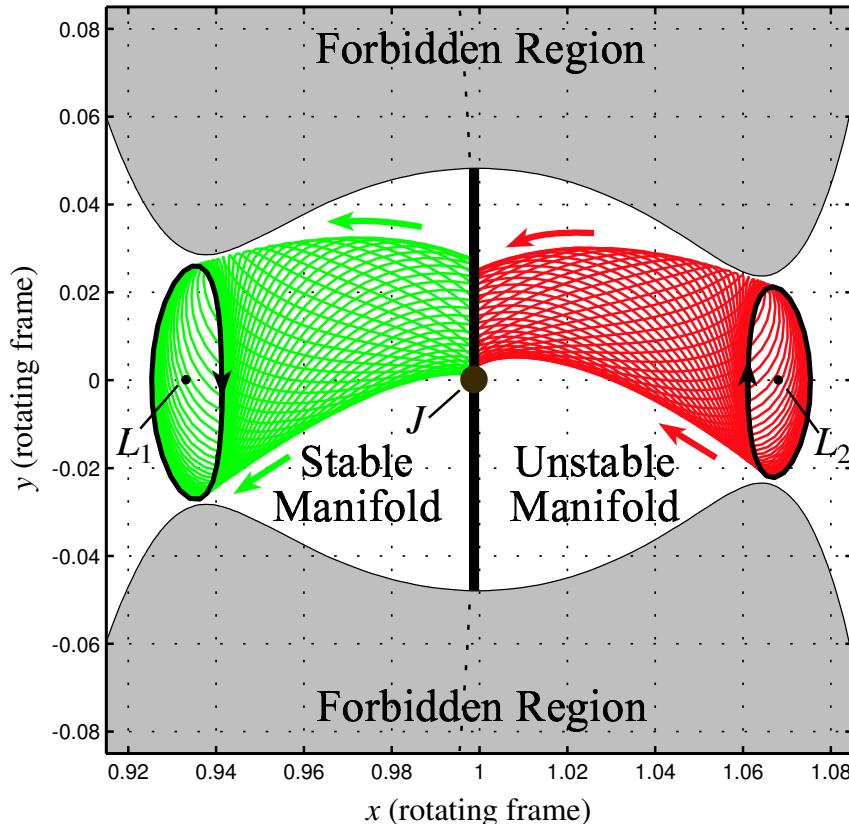
■ Major Result (C): Numerical Construction of Orbits

- Developed procedure to construct orbit with **prescribed itinerary**.
- Example: An orbit with itinerary $(X, J; S, J, X)$.



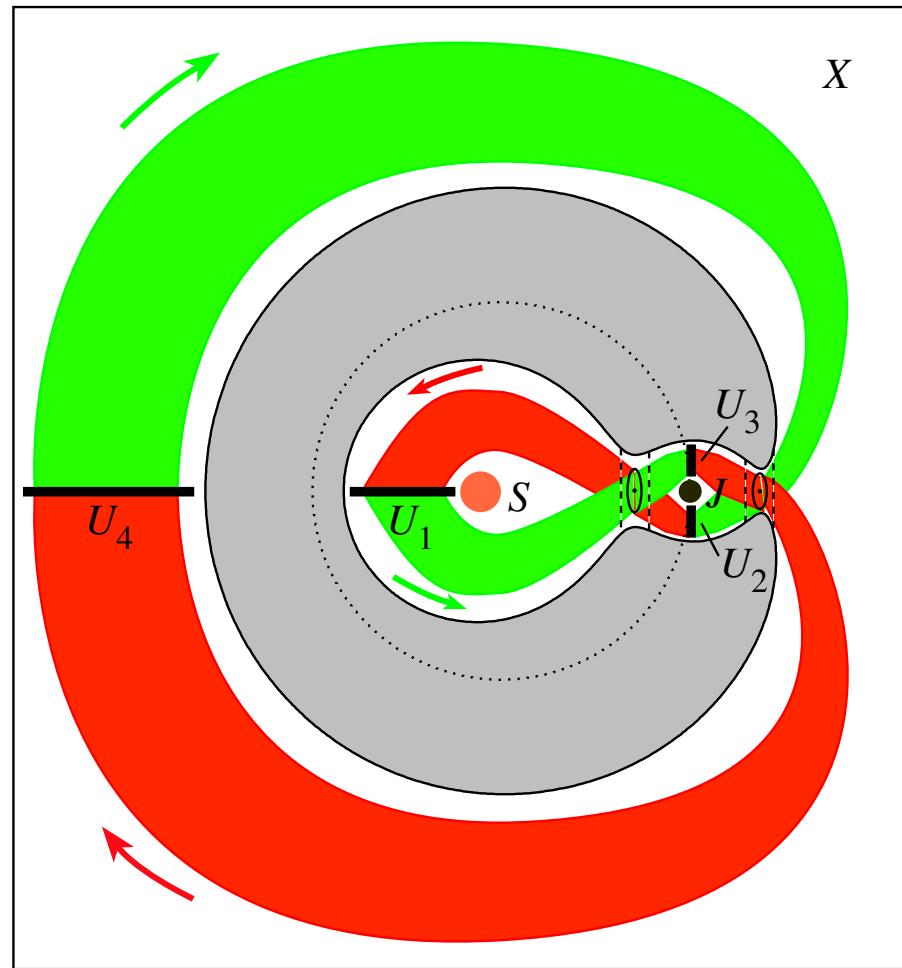
■ Details: Construction of $(\mathbf{J}, \mathbf{X}; \mathbf{J}, \mathbf{S}, \mathbf{J})$ Orbits

- ▶ Invariant manifold **tubes** separate transit from nontransit orbits.
- ▶ **Green curve** (Poincaré cut of L_1 **stable manifold**).
Red curve (cut of L_2 **unstable manifold**).
- ▶ Any point inside the intersection region Δ_J is a $(\mathbf{X}; \mathbf{J}, \mathbf{S})$ orbit.



■ Details: Construction of $(J, X; J, S, J)$ Orbits

- The desired orbit can be constructed by
 - Choosing appropriate **Poincaré sections** and
 - linking invariant **manifold tubes** in right order.



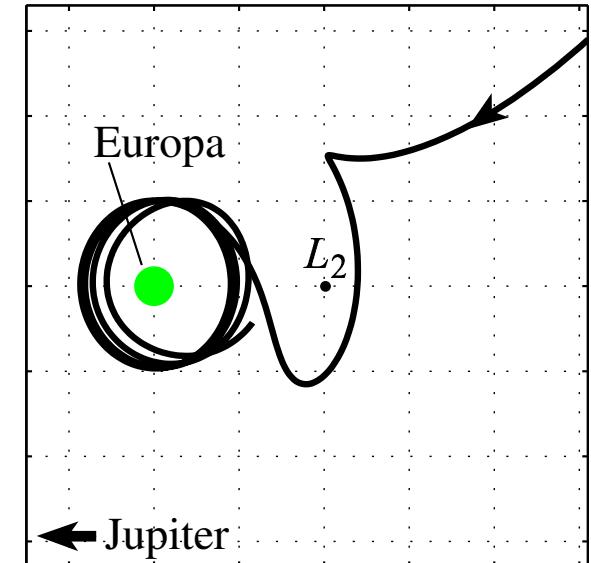
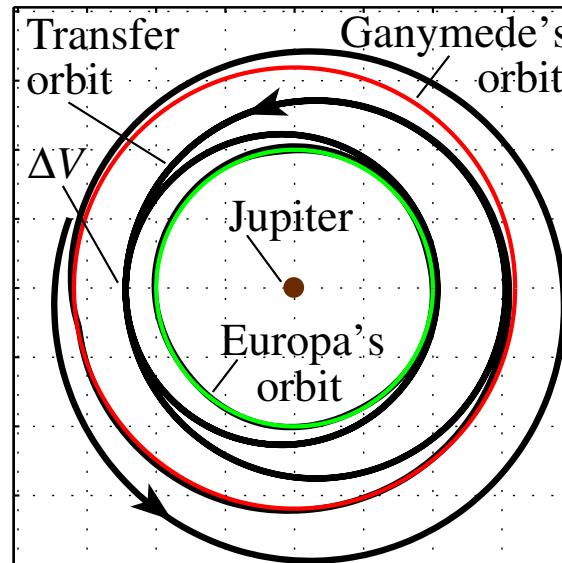
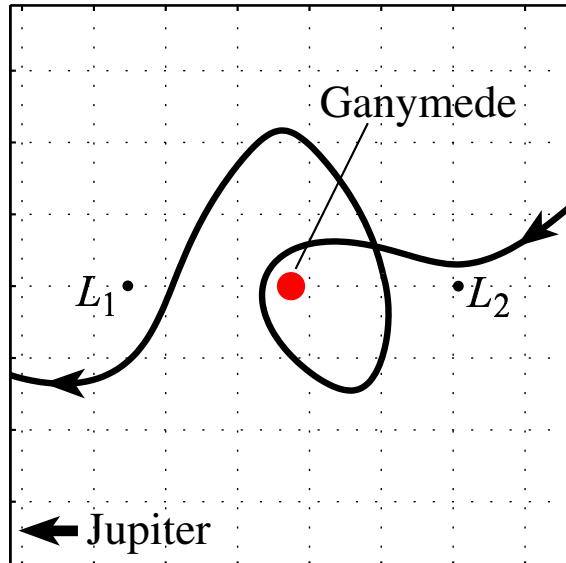
■ Petit Grand Tour of Jupiter's Moons (Planar Model)

► Used invariant manifolds

to construct trajectories with interesting characteristics:

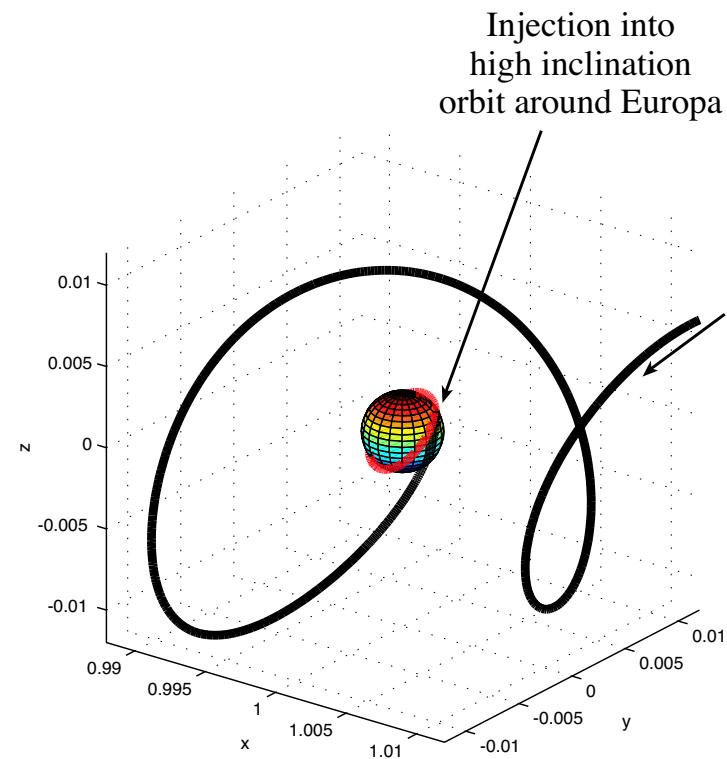
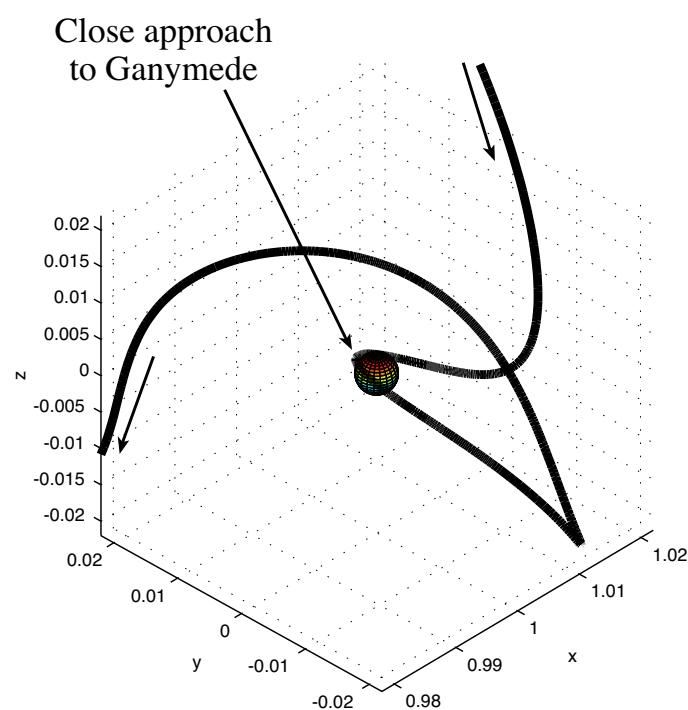
- Petit Grand Tour of Jupiter's moons.
1 orbit around **Ganymede**. 4 orbits around **Europa**.
- A ΔV nudges the SC from
Jupiter-Ganymede system to **Jupiter-Europa** system.

► Instead of **flybys**, can orbit several moons for **any duration**.

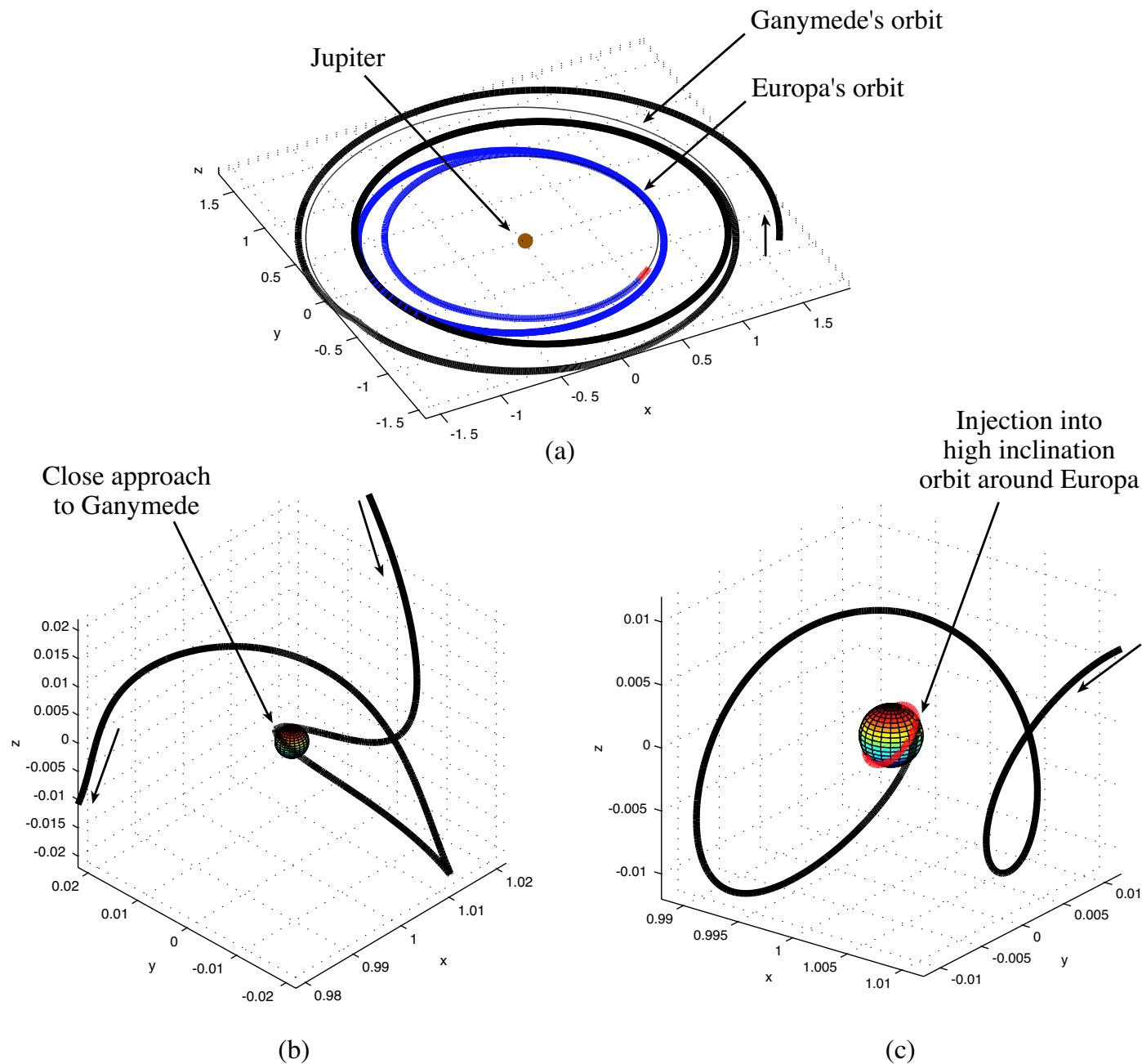


■ Extend from Planar Model to Spatial Model

- ▶ Previous work based on **planar** 3-body problem.
- ▶ Future missions will require **3D** capabilities.
 - Europa Orbiter mission needs a capture into a **high inclination orbit** around Europa.
- ▶ Current study has extended from **planar** to **spatial model**.

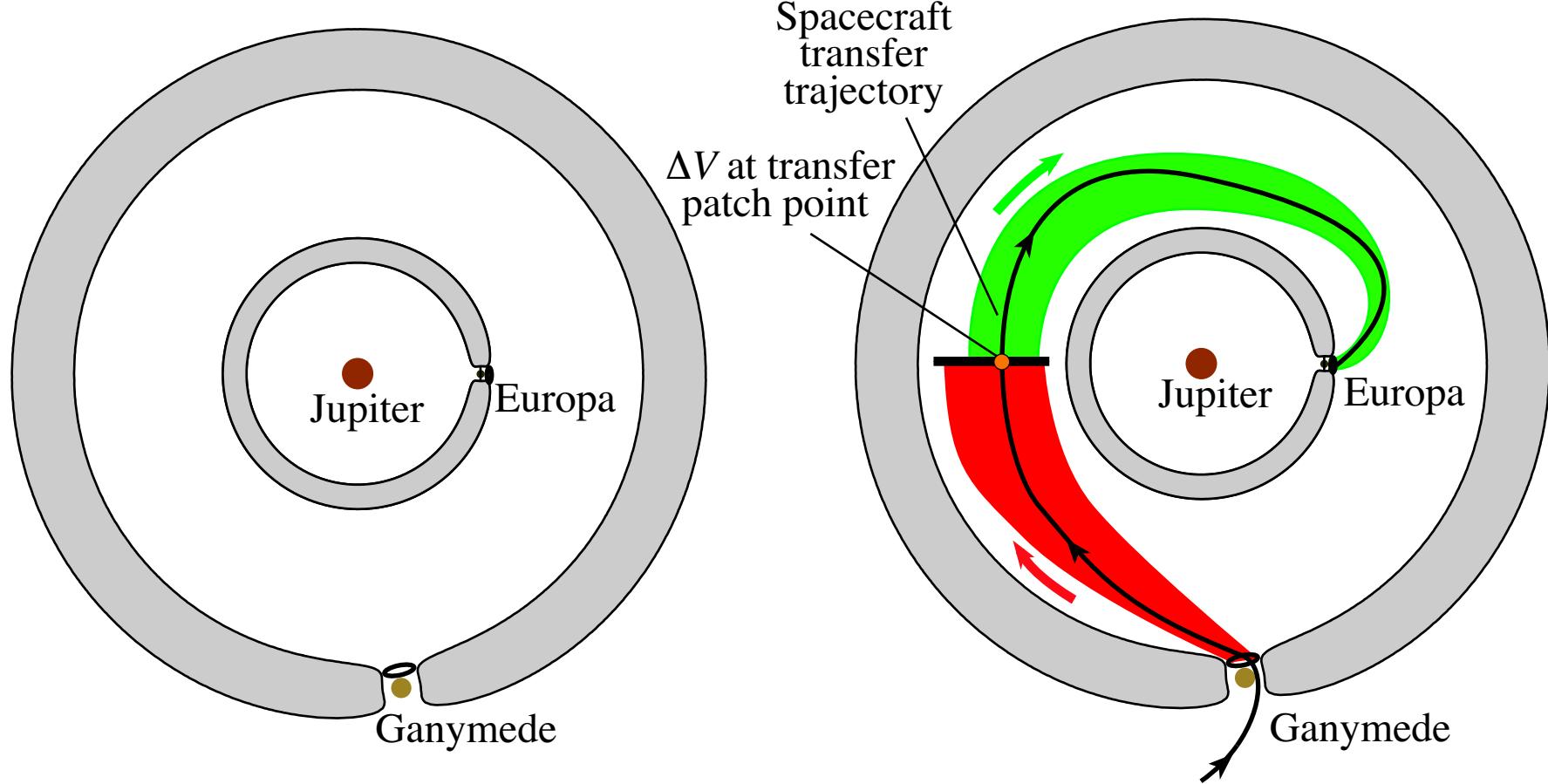


Extend from Planar Model to Spatial Model



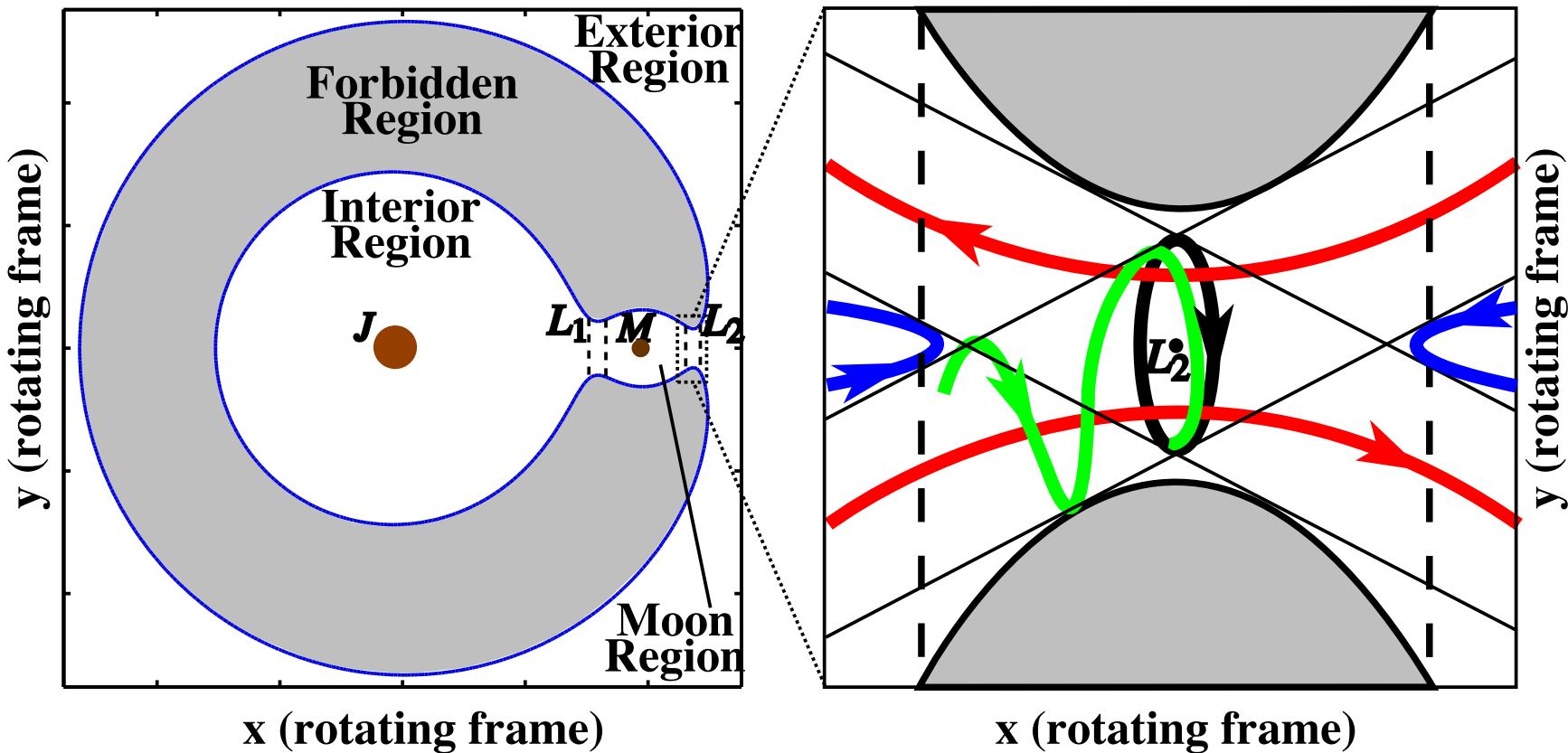
■ Petit Grand Tour of Jupiter's Moons

- Jupiter-Ganymede-Europa-SC 4-body system approximated as 2 **coupled 3-body systems**
- **Invariant manifold tubes** of spatial 3-body systems are linked in right order to construct orbit with desired itinerary.
- Initial solution refined in **4-body model**.



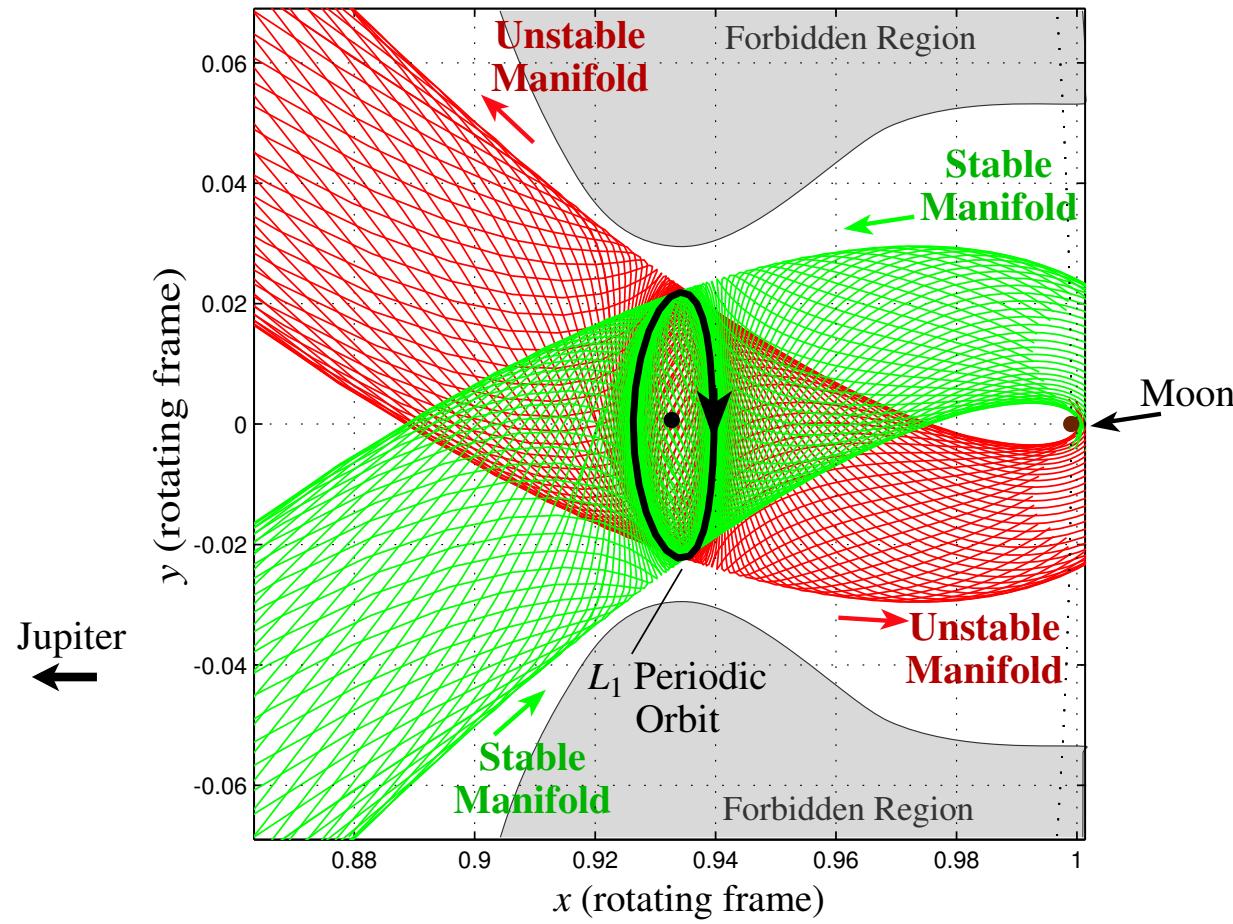
■ Planar Restricted 3-Body Problem

- **Recall:** for energy value just above that of L_2 , Hill's region contains a “neck” about L_1 & L_2 .
- Dynamics in each equilibrium region: **saddle x center**.
- 4 types of orbits:
periodic, asymptotic, transit & nontransit.



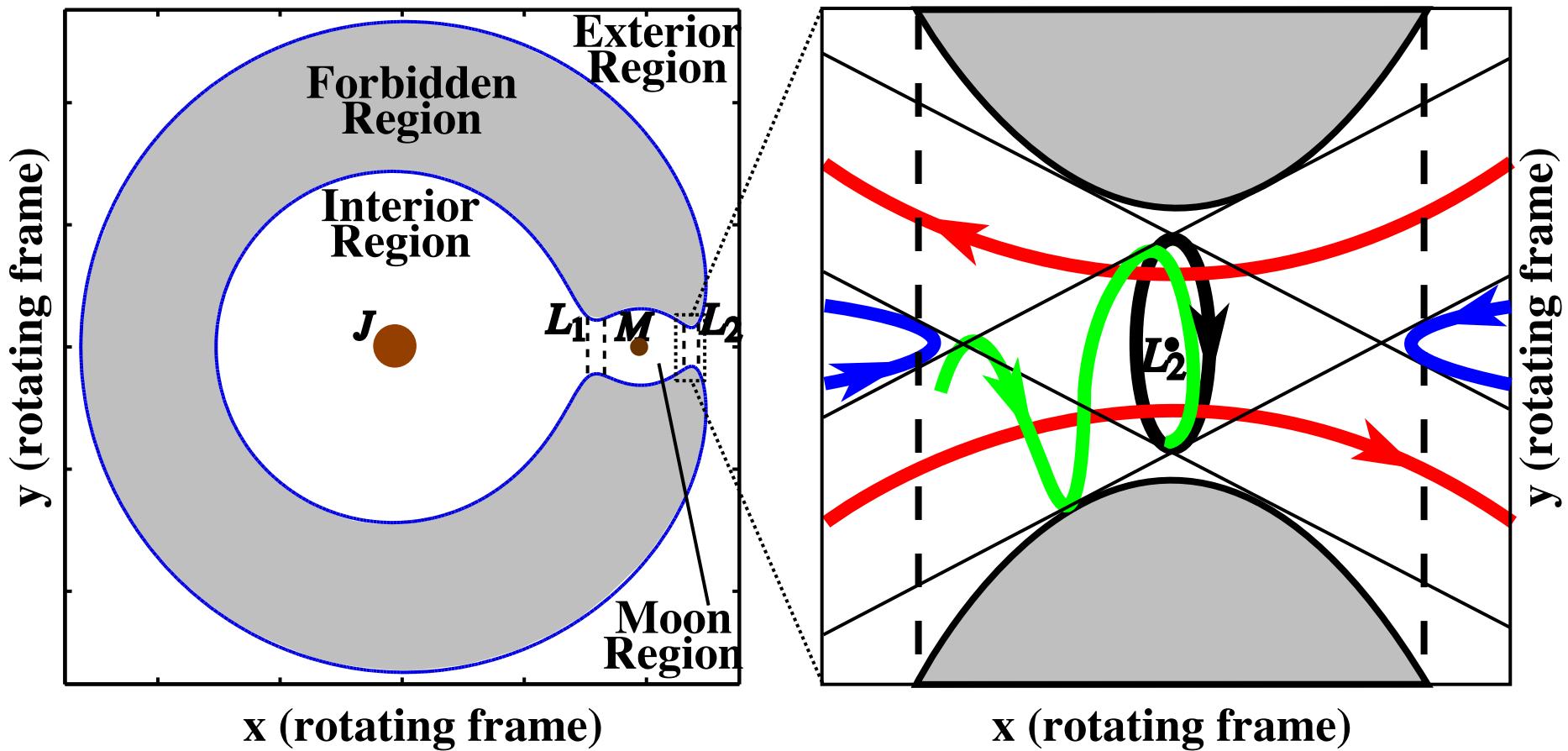
■ Planar: Invariant Manifold as Separatrix

- ▶ Asymptotic orbits form **2D invariant manifold tubes** in **3D energy surface**.
- ▶ They separate transit and non-transit orbits:
 - **Transit orbits** are those inside the tubes.
 - **Non-transit orbits** are those outside the tubes.



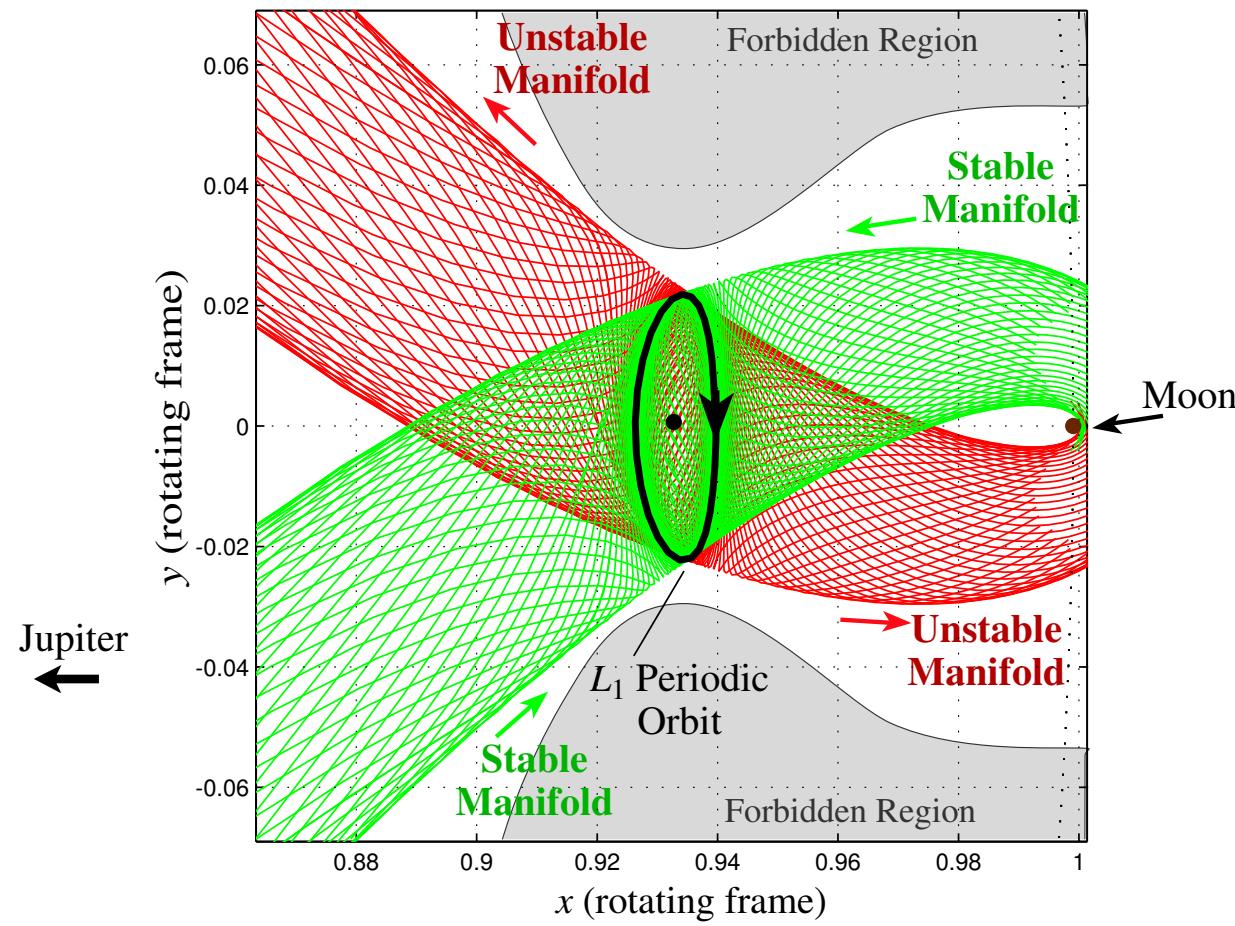
■ Spatial Restricted 3-Body Problem (CR3BP)

- Dynamics near equilibrium point: saddle \times center \times center.
- **bounded orbits** (periodic/quasi-periodic): S^3 (3-sphere)
- **asymptotic orbits** to 3-sphere: $S^3 \times I$ (“tubes”)
- **transit** and **nontransit** orbits.



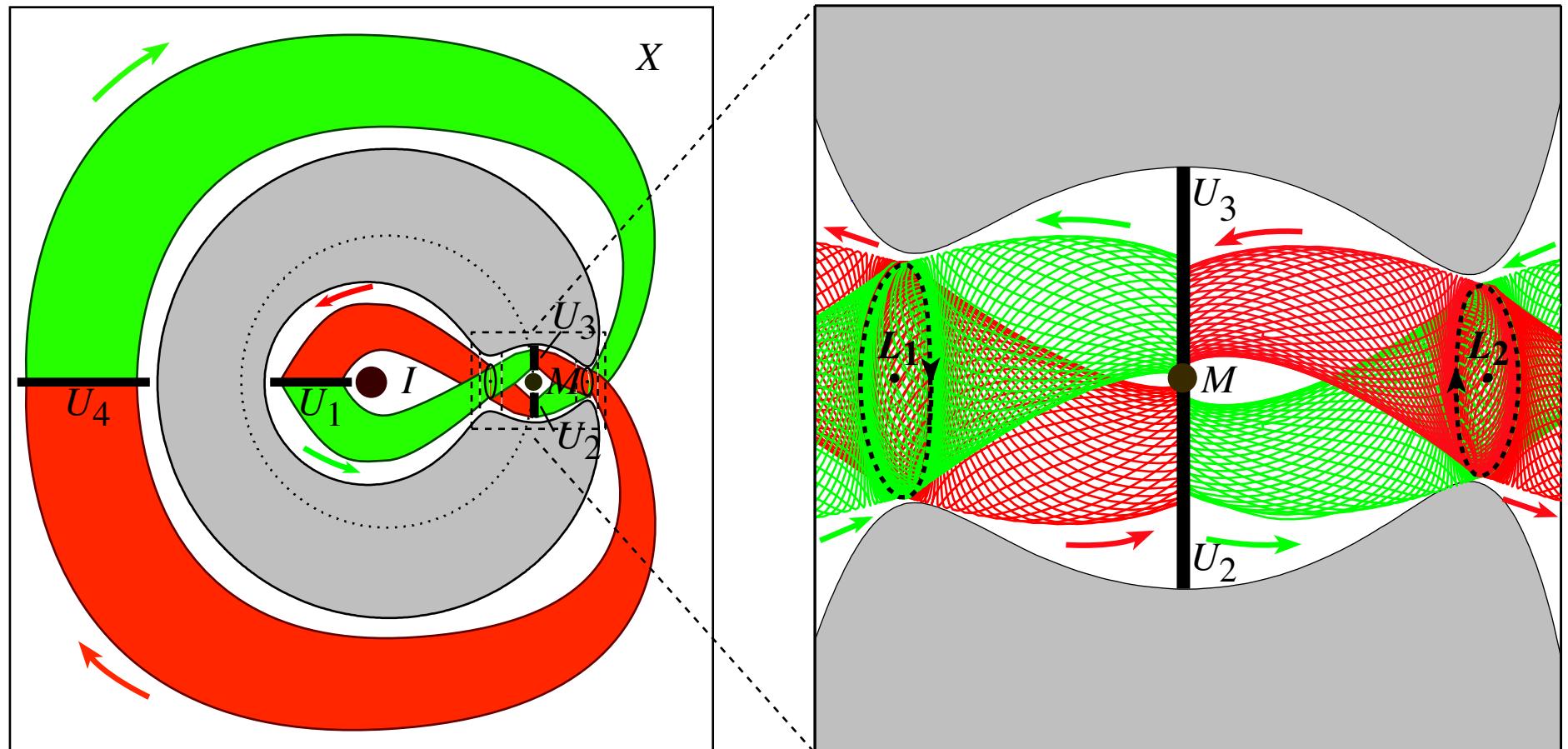
■ Spatial: Invariant Manifold as Separatrix

- ▶ Asymptotic orbits form **4D invariant manifold “tubes”** ($S^3 \times I$) in **5D energy surface**.
- ▶ They separate transit and non-transit orbits:
 - **Transit orbits** are those inside the “tubes”.
 - **Non-transit orbits** are those outside the “tubes”.



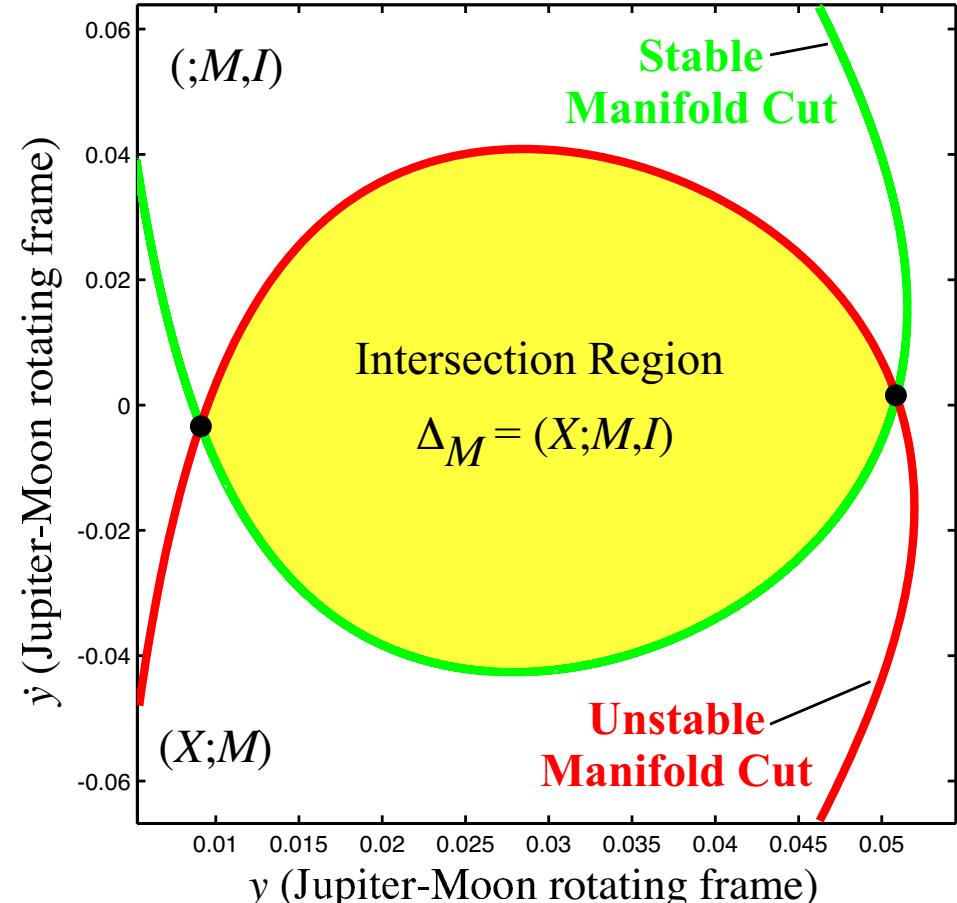
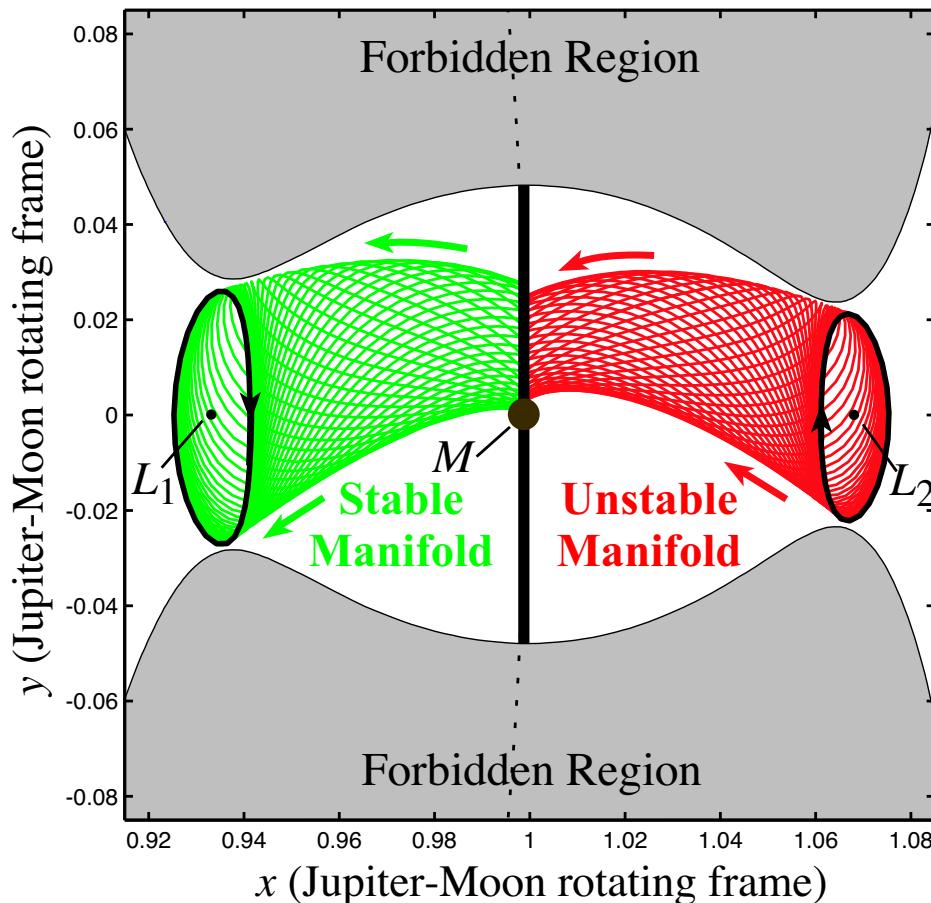
■ Construct Orbit with Desired Itinerary

- ▶ How to link invariant manifold tubes to construct orbit with **desired itinerary**.
- ▶ Construction of $(X; M, I)$ orbit.



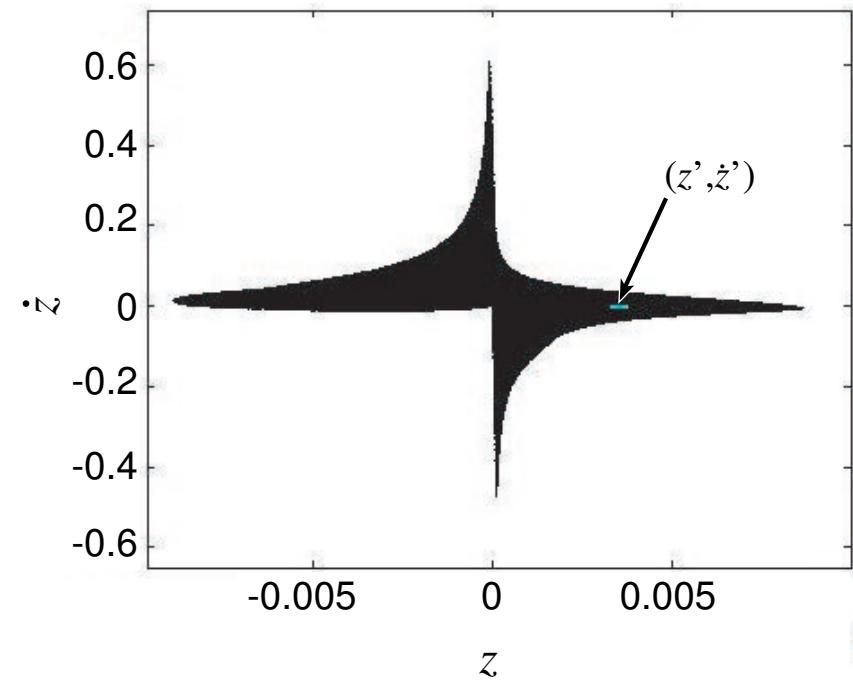
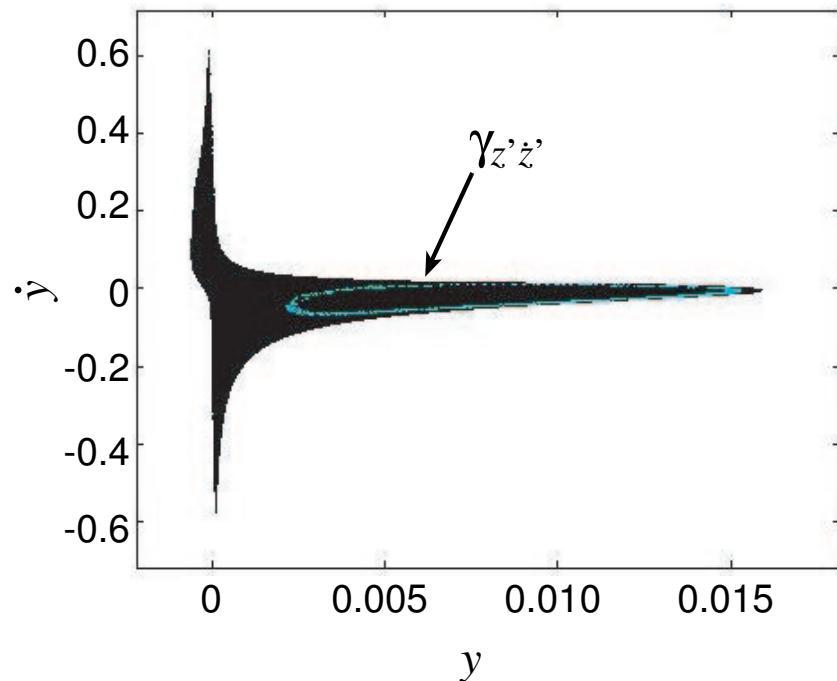
■ Planer: Construction of $(X; M, I)$ Orbits

- Invariant mfd. **tubes** ($S \times I$) separate transit/nontransit orbits.
- **Red curve (S^1)** (Poincaré cut of L_2 **unstable manifold**).
Green curve (S^1) (cut of L_1 **stable manifold**).
- Any point inside the intersection region Δ_M is a $(X; M, I)$ orbit.



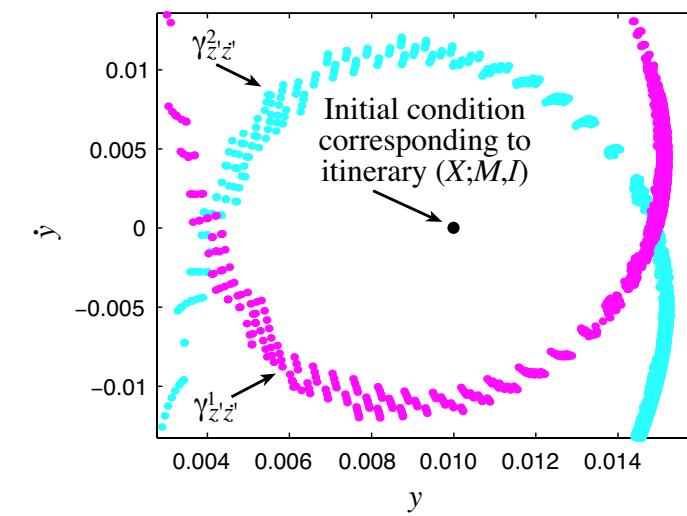
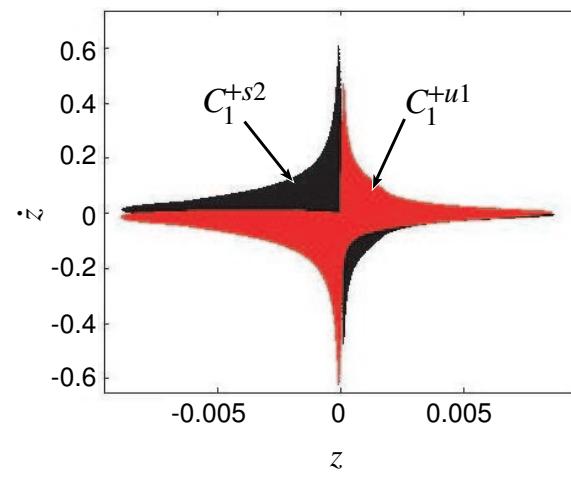
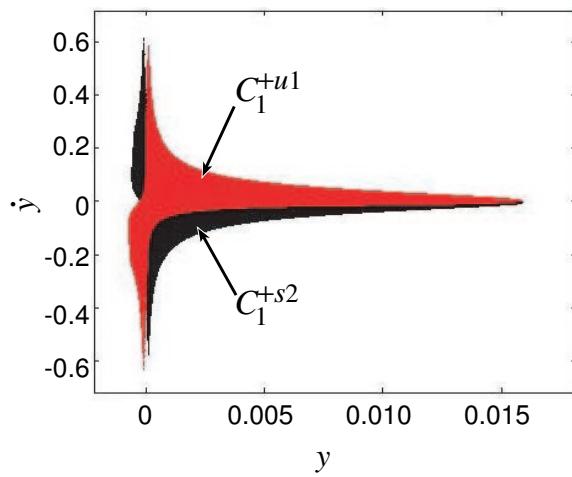
■ Spatial: Construction of $(X; M, I)$ orbits

- ▶ Invariant manifold **tubes**: $(S^3 \times I)$.
- ▶ Poincaré **cut** is a topological **3-sphere** S^3 in \mathbb{R}^4 .
 - S^3 looks like **disk x disk**: $\xi^2 + \dot{\xi}^2 + \eta^2 + \dot{\eta}^2 = r^2 = r_\xi^2 + r_\eta^2$
- ▶ If $z = c, \dot{z} = 0$, its **projection** on (y, \dot{y}) **plane** is a **curve**.
- ▶ Any point inside this **curve** is a $(X; M)$ orbit.

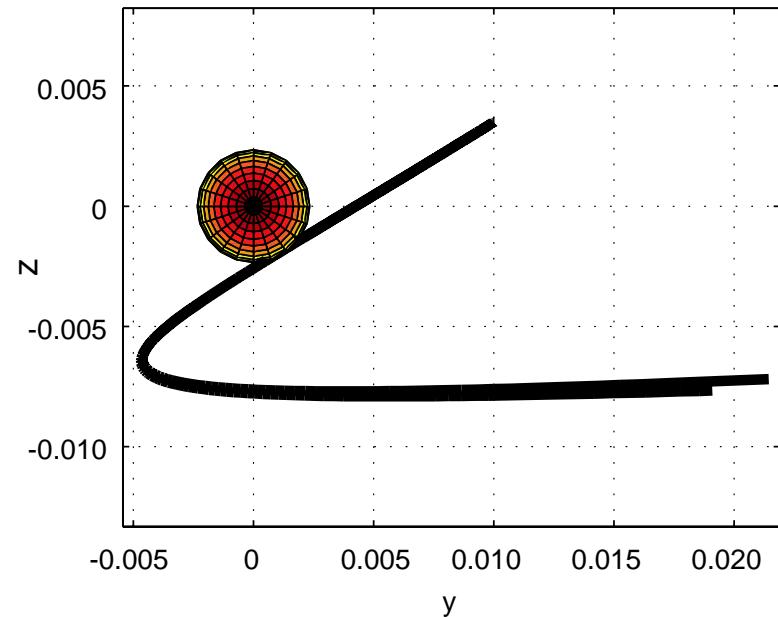
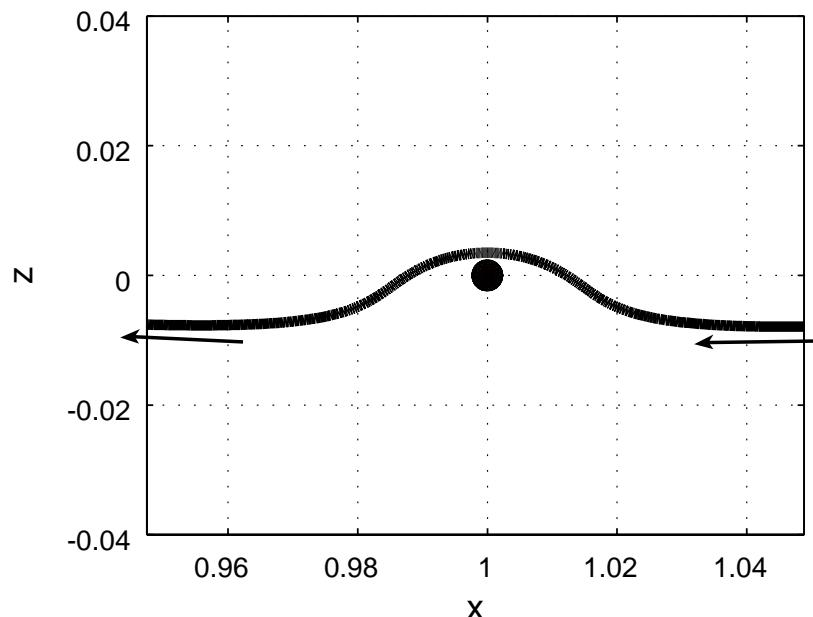
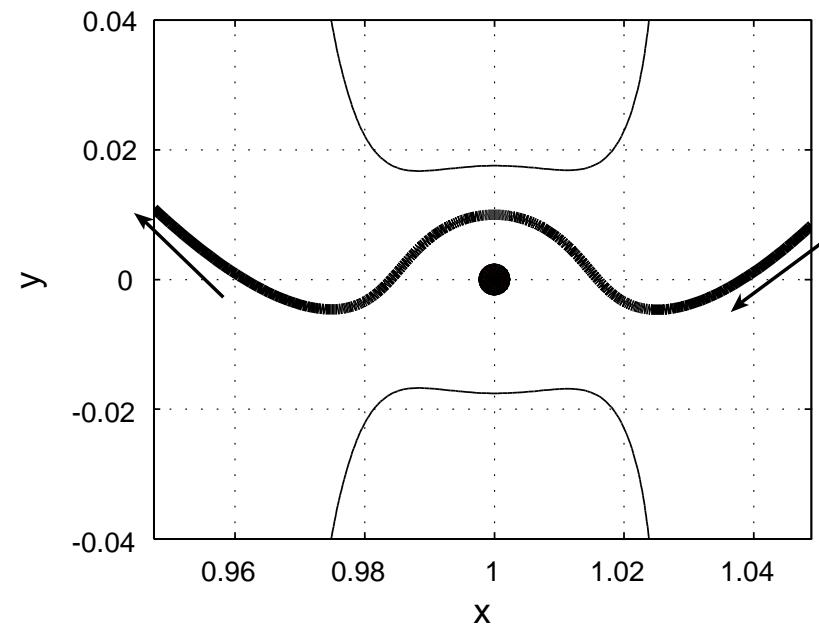
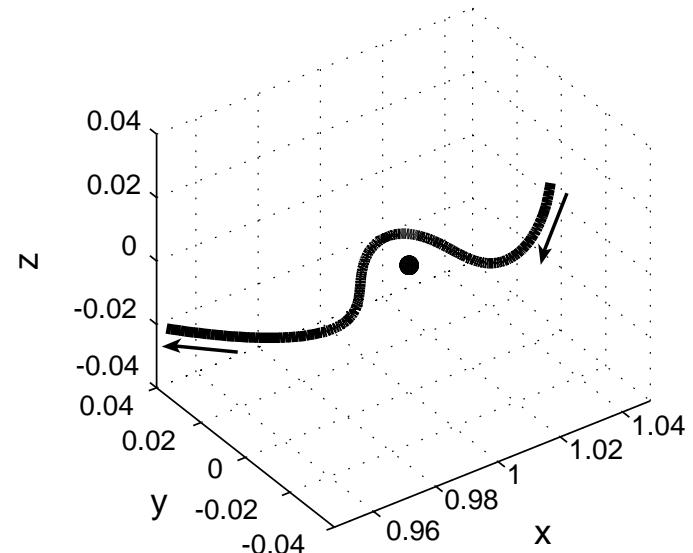


■ Spatial: Construction of $(X; M, I)$ orbits

- ▶ Similarly, while **cut** of stable manifold “**tube**” is a S^3 , its **projection** on (y, \dot{y}) **plane** is a **curve** for $z = c, \dot{z} = 0$.
- ▶ Any point inside this **curve** is a (M, I) orbit.
- ▶ Hence, any point inside the **intersection region** Δ_M is a $(X; M, I)$ orbit.

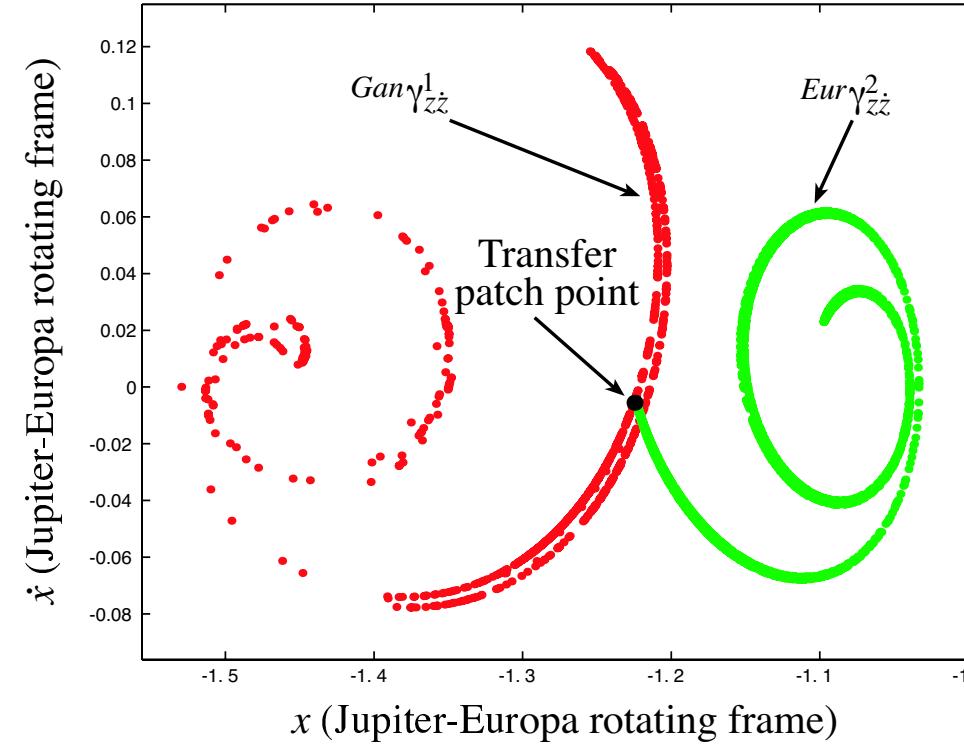
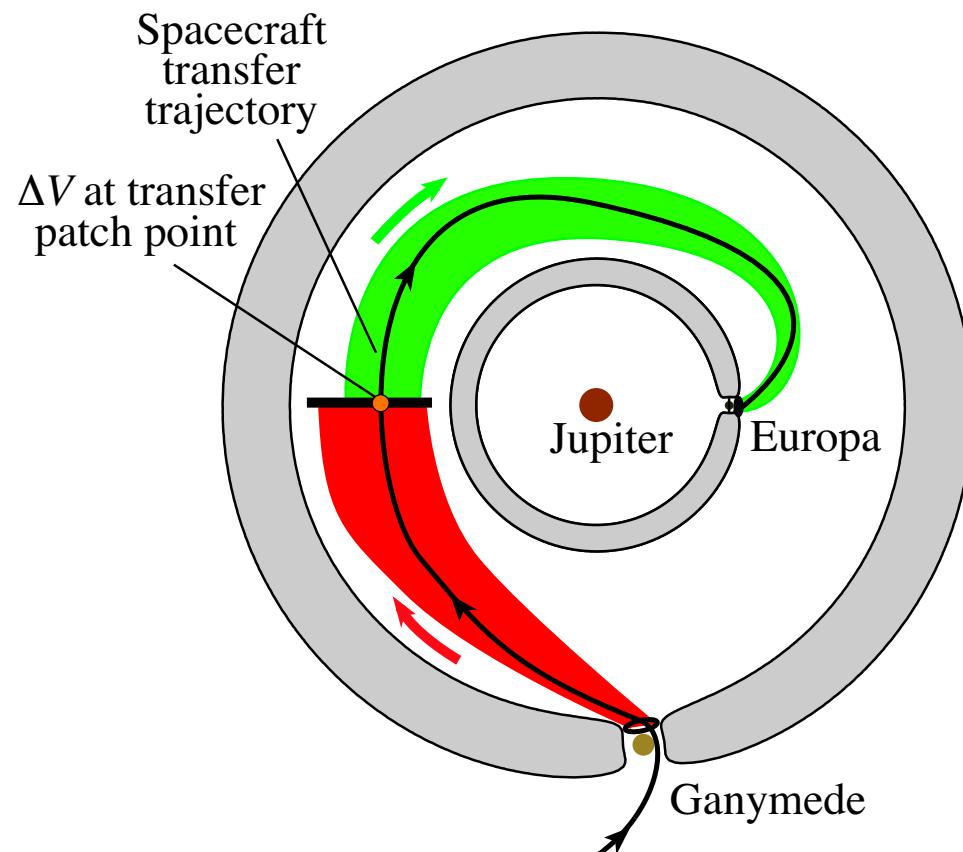


■ Spatial: Construction of $(X; M, I)$ orbits



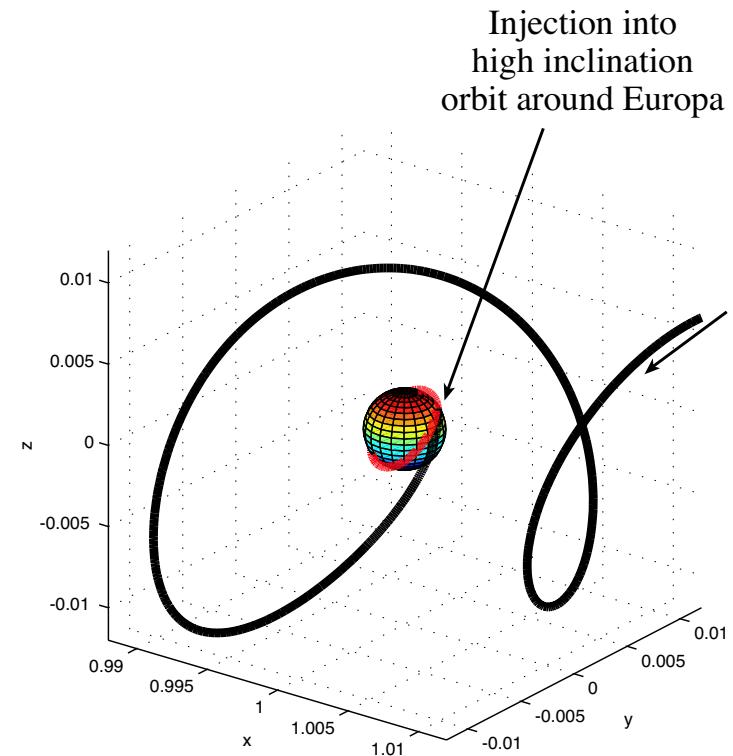
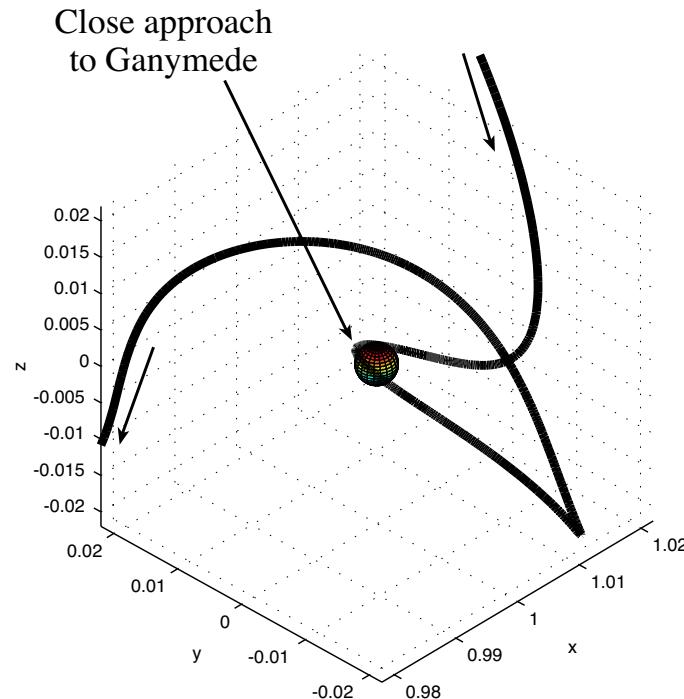
Petit Grand Tour of Jupiter's Moon

- Petit Grand Tour can be constructed similarly
 - Approximate 4-body system as 2 nested **3-body systems**.
 - Choose appropriate **Poincaré section**.
 - Link invariant **manifold tubes** in right order.
 - Refine initial solution in **4-body model**.



■ Conclusion and Future Work

- ▶ In our study of **spatial CR3BP**:
 - Invariant manifolds still act as **separtrices**.
 - Construct orbit with **prescribed itinerary**.
 - Construct **Petit Grand Tour** of Jupiter's moons that ends in a **high inclination** orbit around Europa.
- ▶ Will construct useful trajectories in **Sun-Earth-Moon** system.

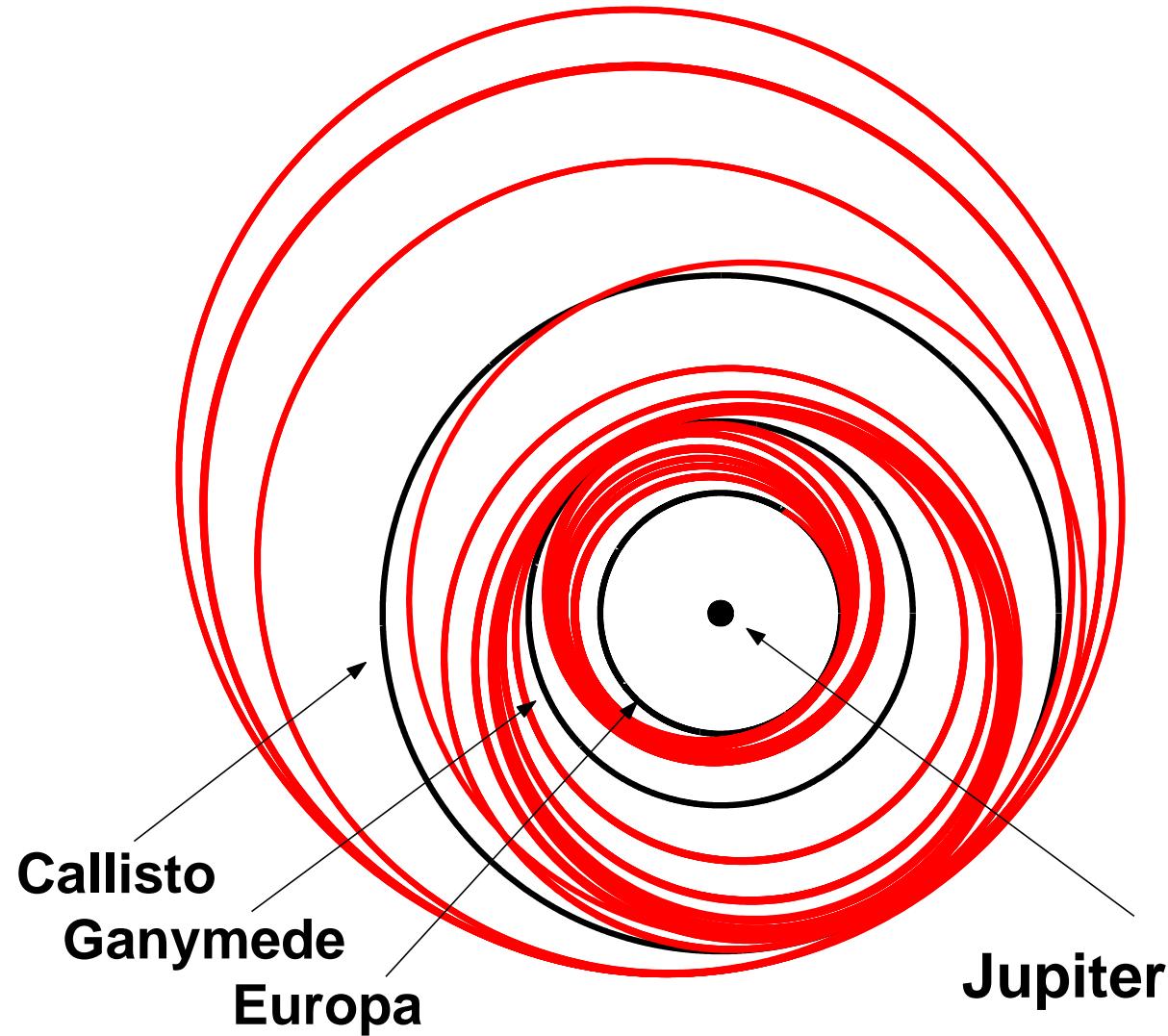


■ Future Work: Pump Down via Resonance Encounters

- For this new tour: $\Delta V = 20m/s$.

Low Energy Tour of Jupiter's Moons

Seen in Jovicentric Inertial Frame



■ References and Other Informations

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 - <http://www.cds.caltech.edu/~koon/>
 - <http://ojps.aip.org/chaos/>
- ▶ **Gómez, G., W.S. Koon, M.W. Lo, J.E. Marsden, J. Masdemont and S.D. Ross**
Invariant Manifolds, The Spatial Three-Body Problem and Space Mission Design.
- ▶ Email: koon@cds.caltech.edu